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NIKE CAJUN SOUNDING ROCKET PERFORMANCE

OTS PRICE

~~H. H. Lane~~
~~42p. orig.~~
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MICROFILM \$

PREPARED BY
JOHN H. LANE and
NANCY L. KEMMERER ap. 1962

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APRIL 1962

NASA

GODDARD SPACE FLIGHT CENTER
GREENBELT, MD.

NIKE CAJUN SOUNDING ROCKET

PERFORMANCE

DATE: April 17, 1962

TO : Files
FROM : Code 616.2 - John H. Lane and Nancy L. Kemmerer
SUBJECT : Nike-Cajun Sounding Rocket Performance

The Nike-Cajun (Figure 1) is the most widely used vehicle in the NASA National Sounding Rocket Program. It has also seen extensive service with other U.S. research agencies and with nations overseas.

Goddard's ability to accurately predict Nike-Cajun performance has increased significantly within the last year as a result of:

1. Completion of trajectory studies run on an IBM 7090 computer,
2. Analysis of Nike-Cajun radar data and comparison of theoretical and actual performance,
3. Preparation of curves for predicting performance of the Nike-Cajun system.

Because Nike-Cajun performance is highly sensitive to variations in Cajun drag, it was necessary to run calculations with what may be termed "low" "medium", and "high" drag coefficient inputs. The Cajun drag configurations considered were:

CASE I

Low Drag: Clean payload (no antennae) with 11° total angle (6.75 inch diameter) nose cone (see Cajun drawing, Fig. 2).

CASE II

Medium Drag: Two dovap antennae (Fig. 3) with 11° total angle (6.75 inch diameter) nose cone

or

Medium Drag: Clean payload (no antennae) with 20° total angle (6.75 inch diameter) nose cone.

CASE III

High Drag: Four 45° swept turnstile antennae (Fig. 4) with 11° total angle (6.75 inch diameter) nose cone.

Computer calculations were conducted for Nike-Cajun Rockets having the following characteristics:

1. Sea level launch altitude
2. Launch elevation angles of 70, 75, 80, 82, 84, 86, and 90 degrees
3. Gross payload weights of 40, 50, 60, 70, 80, 90, 100, 110, and 120 lbs. (gross payload defined as all weight forward of the Cajun motor head).
4. Cajun ignition time of 17 seconds
5. Cajun drag configuration: Case I, Case II, Case III.

Additional information on assumptions and inputs for these calculations are given in Table 1 and Figures 5 through 8.

Results of the Nike-Cajun calculations are listed in Tables 2, 3, and 4. Major performance parameters at Nike burnout and Cajun ignition, burnout, apogee, and impact are plotted as a function of payload and launch angle in Figures 9 through 23. These curves are intended for use in predicting the performance of the Nike-Cajun having drag coefficient curves identical or similar to those of drag cases I, II, III.

A detailed evaluation of available Nike-Cajun radar data is now being conducted to determine adequacy of inputs to the theoretical calculations. Partial results, a comparison of theoretical and actual (radar) performance for eight rockets, are shown in Table 5. Here, the average discrepancies in

peak altitude and impact range were found to be only 3.5 and 2.9 statute miles respectively excluding results of NASA 10.74 which is known to have experienced pitch-roll coupling (roll lock-in) shortly after Cajun burnout. It should be emphasized that the theoretical values shown in Table 5 result from post-flight calculations using actual (radar) values for Cajun ignition time and effective flight path angle.

The accuracy of pre-flight estimates are often adversely affected by limitations of the firing range in predicting wind effects and, hence, effective launch elevation angle. For a typical Nike-Cajun, a 2° discrepancy in effective angle results in a peak altitude error of 2.5 miles and an impact range error of about 15 miles. It appears that, in general, peak altitude predictions within plus and minus six miles of actual can be expected if a very careful determination of weight and drag configuration is made.

Performance investigations have not yet been run for some Cajun pay-load geometries. The "grenade series" (4 dovap antennae and an 18 inch long nose cone with 100 lbs of extra weight in the Nike adapter) is in this category.

RECOMMENDATIONS FOR USE OF NIKE-CAJUN PERFORMANCE CURVES

1. Carefully determine the correct drag case, payload weight, and launch angle (desired effective launch angle) before entering curves.
2. Specify at least a plus-and-minus six mile tolerance on peak altitude predictions.

3. Consult with Code 616.2 in situations where drag configuration does not appear to match any of the three cases defined above, or where other deviations from normal Nike-Cajun input parameters are present.

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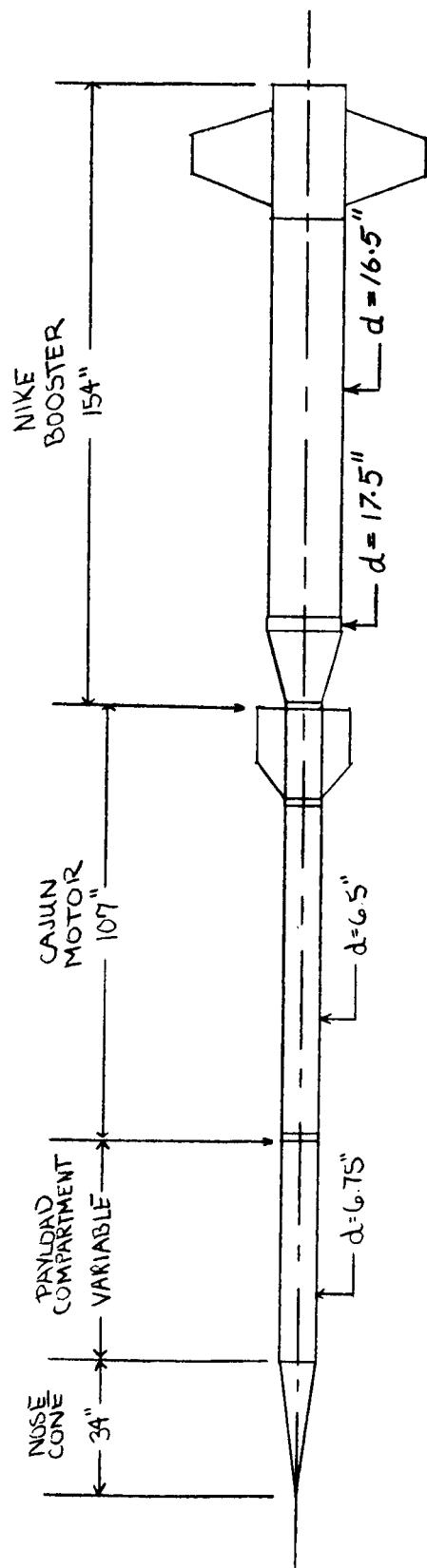
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NIKE CAJUN

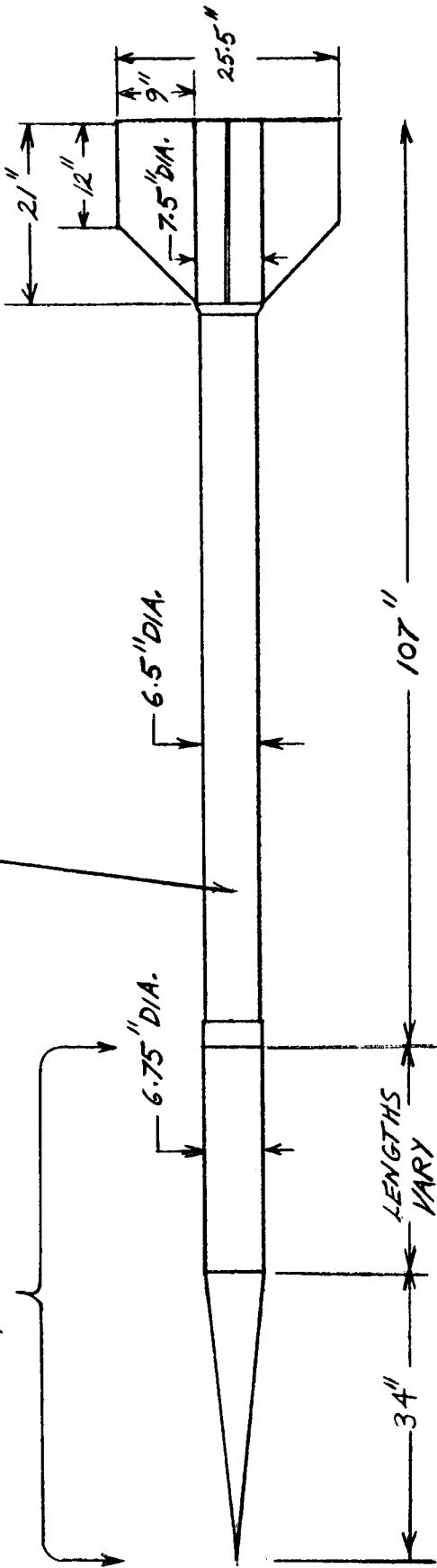


(LENGTHS TO NEAREST INCH)

FIGURE 1

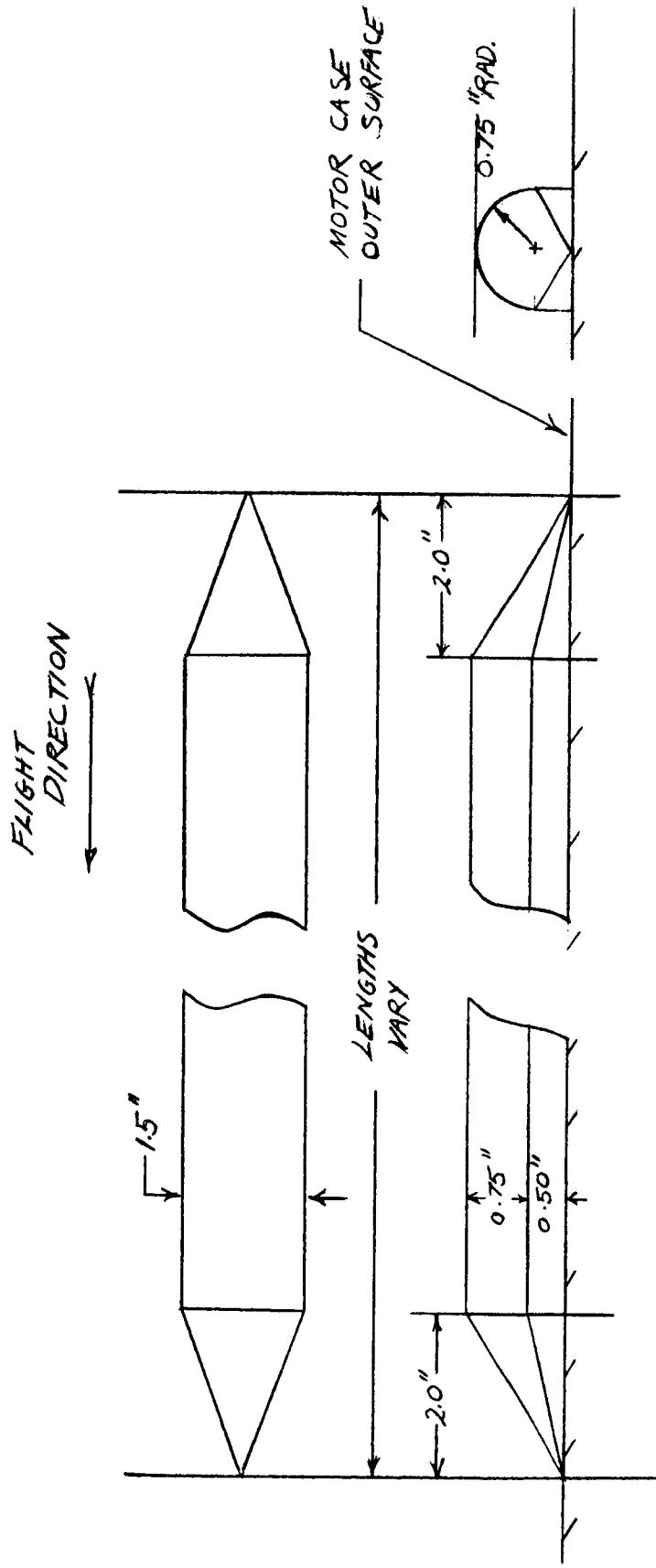
PAYOUT SECTION
SHOWN WITH 11° TOTAL ANGLE
NOSE CONE, NO ANTENNAE

CAJUN MOTOR



CAJUN ROCKET

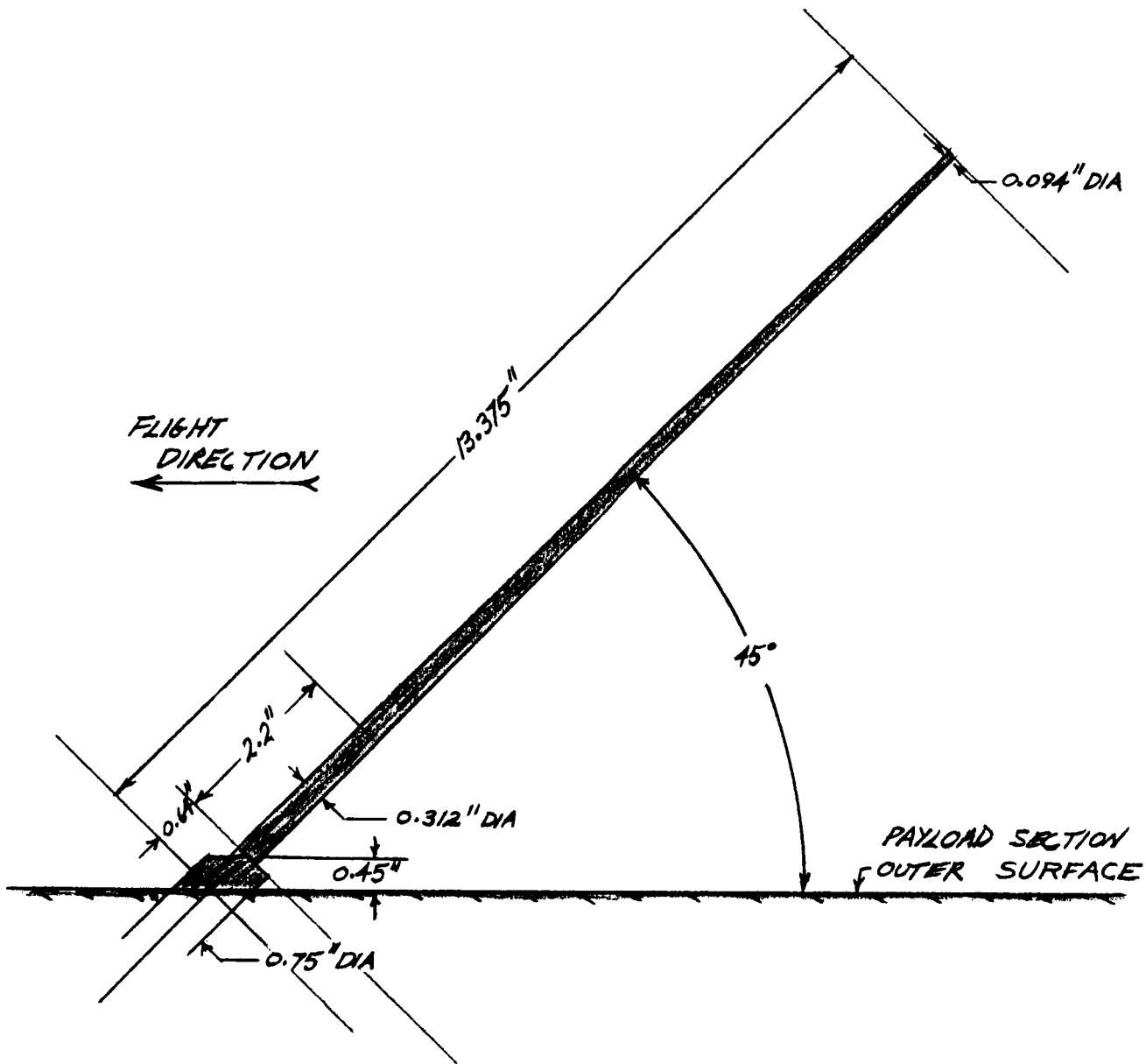
FIGURE 2



DOVAP ANTENNA

JML MAR 62

FIGURE 3



45° SWEPT TURNSTILE ANTENNA

TABLE I

ASSUMPTIONS AND INPUTS FOR NIKE-CAJUN TRAJECTORY CALCULATIONS

ASSUMPTIONS

1. Two dimensional point-mass trajectory.
2. Spherical, non-rotating Earth.
3. Inverse square gravitational field.
4. Standard Atmosphere, ICAO extension.
5. Sea level launch.
6. 17 second Cajun ignition.

INPUTSMass

$w = w + w_{P/L} = \text{total mass (lbs. mass)}$ where w is mass without payload;
 $w_{P/L}$ is payload mass.

At $t = 0$ sec. (Nike Ignition) $w = 1520$ lbs. mass

At $t = 3.5$ sec. (Nike Burnout before separation) $w = 765$ lbs. mass.

From 0 to 3.5 sec. mass flow rate is assumed constant,

$$\dot{w} = \frac{755}{3.5} = 215.714 \text{ lbs. mass/sec}$$

From 3.5 seconds (after stage separation) to 17.0 seconds, $w = 202$ lbs. mass.

From 17.0 to 21.0 seconds (Cajun burning) w is as given in figure 5.

From 21.0 seconds to impact, $w = 83$ lbs. mass.

Payload weights, $w_{P/L}$ are taken at 10 lbs. mass intervals from 40 to 120 lbs. mass.

Thrust

$$T = T_{SL} + A_E (P_E - P_A)$$

where

T = total thrust (lbs)

T_{SL} = thrust at sea level (lbs)

A_E = nozzle exit area (square inches)

P_E = nozzle exit pressure = 14.7 psia

P_A = ambient pressure (psia)

Nike Burning:

Burning period : 0 to 3.5 seconds

$T_{SL} = 42,500$ lbs. = constant

$$A_E = \frac{\pi}{4} (16.333)^2 = 209.4 \text{ inches}^2$$

TABLE I Cont.

Cajun Burning:

Burning period: 17.0 to 21.0 seconds
 T_{SL} = as per figure 6
 A_E = 37.1 inches²

Drag

Drag coefficients for Nike thrusting period as given in figure 7.

Drag coefficients for Cajun coasting and thrusting periods as given in figure 8 (for Cajun payload configuration I, II, and III).

Launch Elevation Angles

Taken as 70°, 75°, 80°, 82°, 84°, 86°, and 90°.

CALCULATED
LOAD VS. TIME

200

180

160

140

120

100

80

60

40

20

0

MASS (LBS. MASS)

50

45

40

35

30

25

20

15

10

5

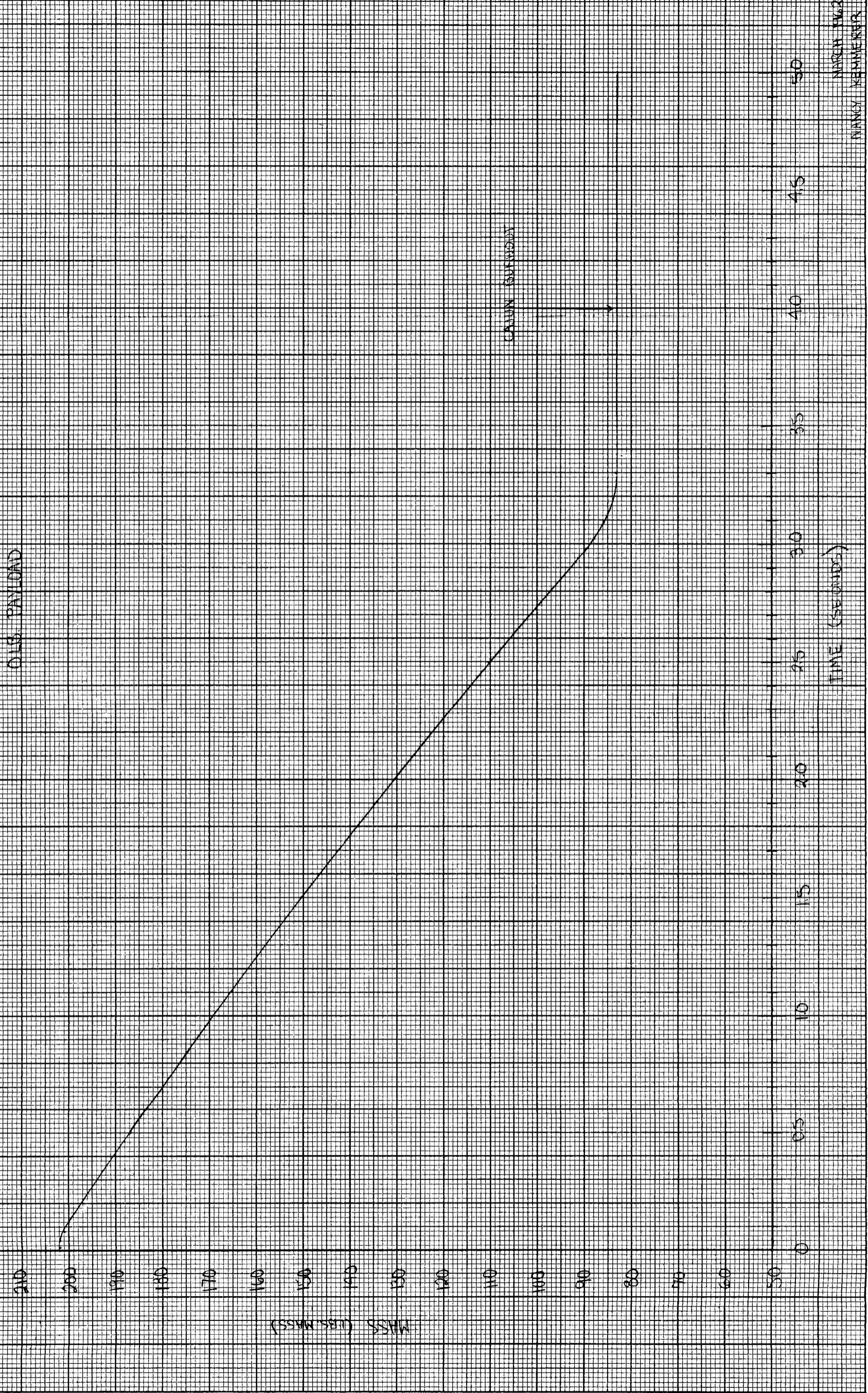
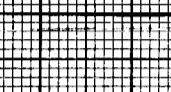
0

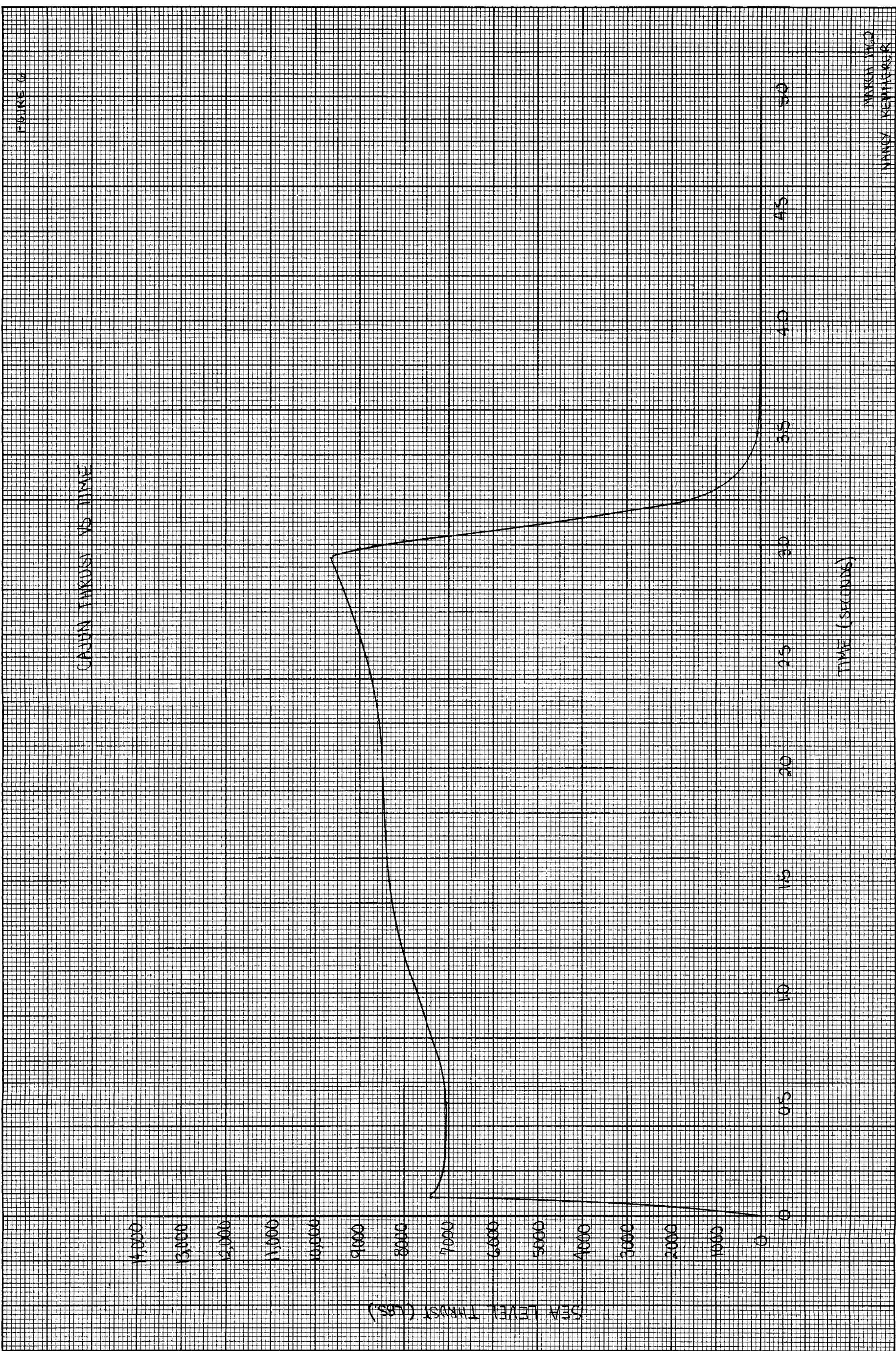
TIME (SECONDS)

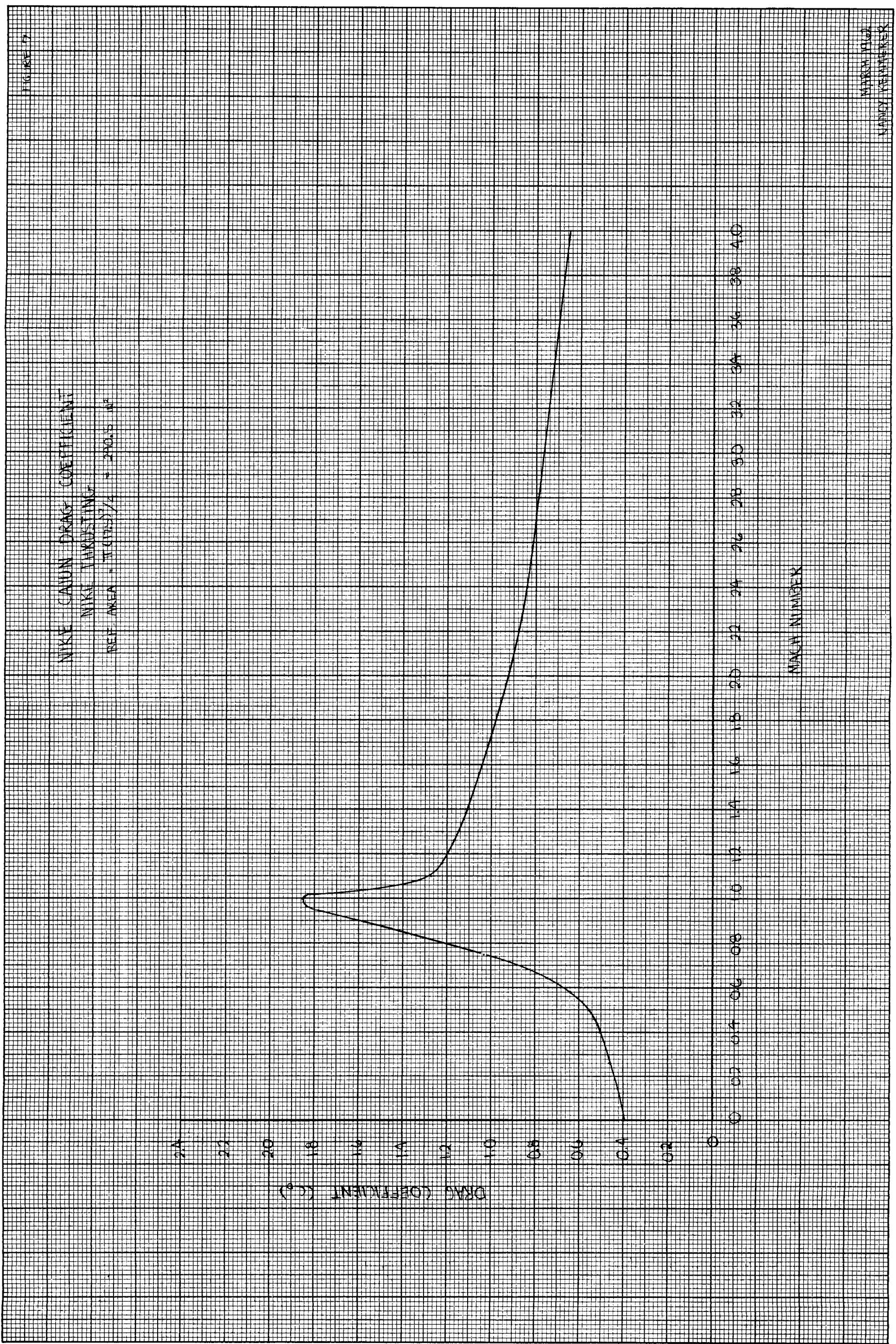
TEST NO. 2
NARROW

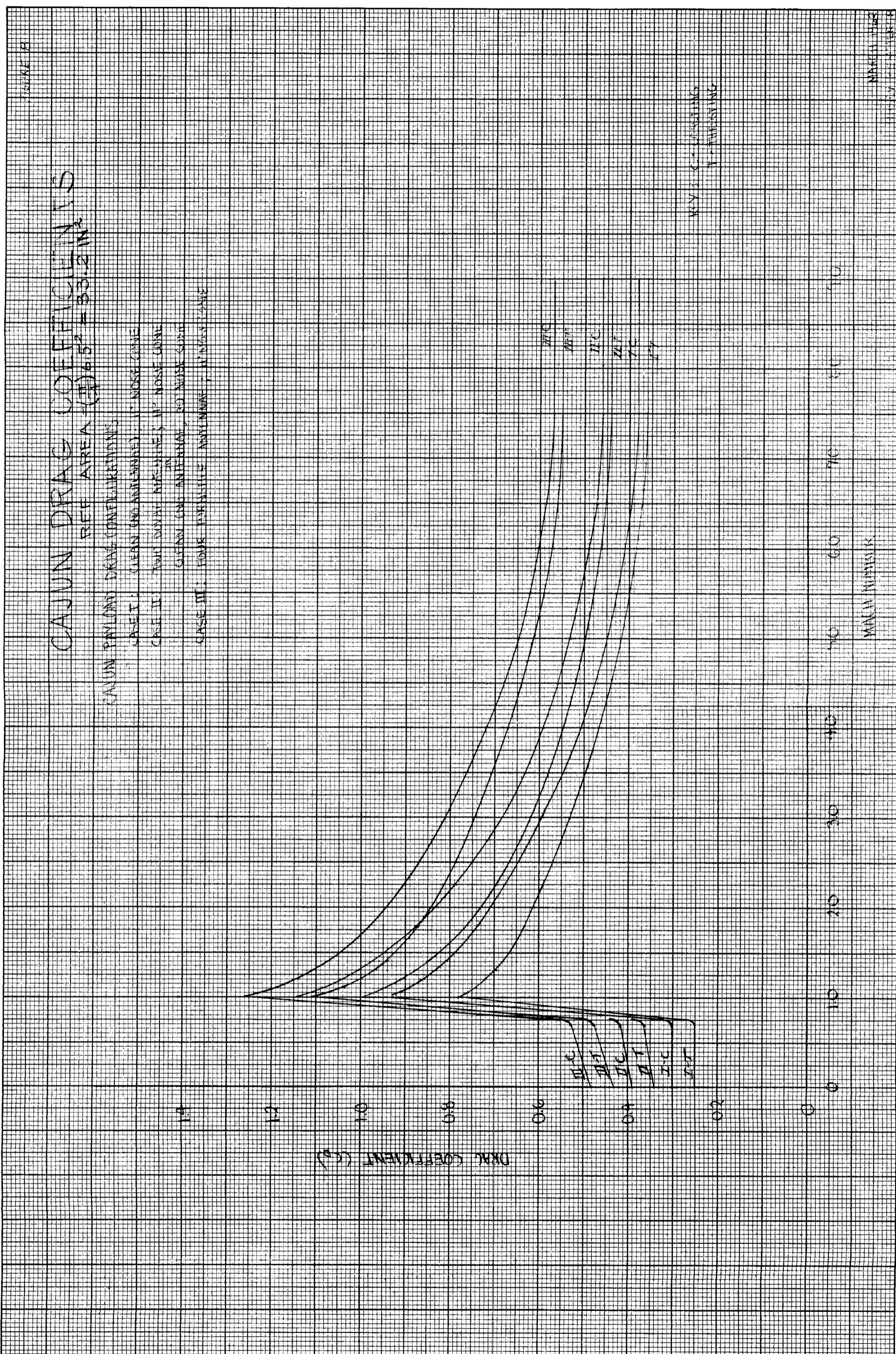
TEST NO. 1

DOWN STROKE









CASE I DRAG CONFIGURATION

CLEAN (NO ANTENNAE) WITH 11° NOSE CONE

TABLE 2

THEORETICAL NIKE CAJUN - PERFORMANCE
CASE I

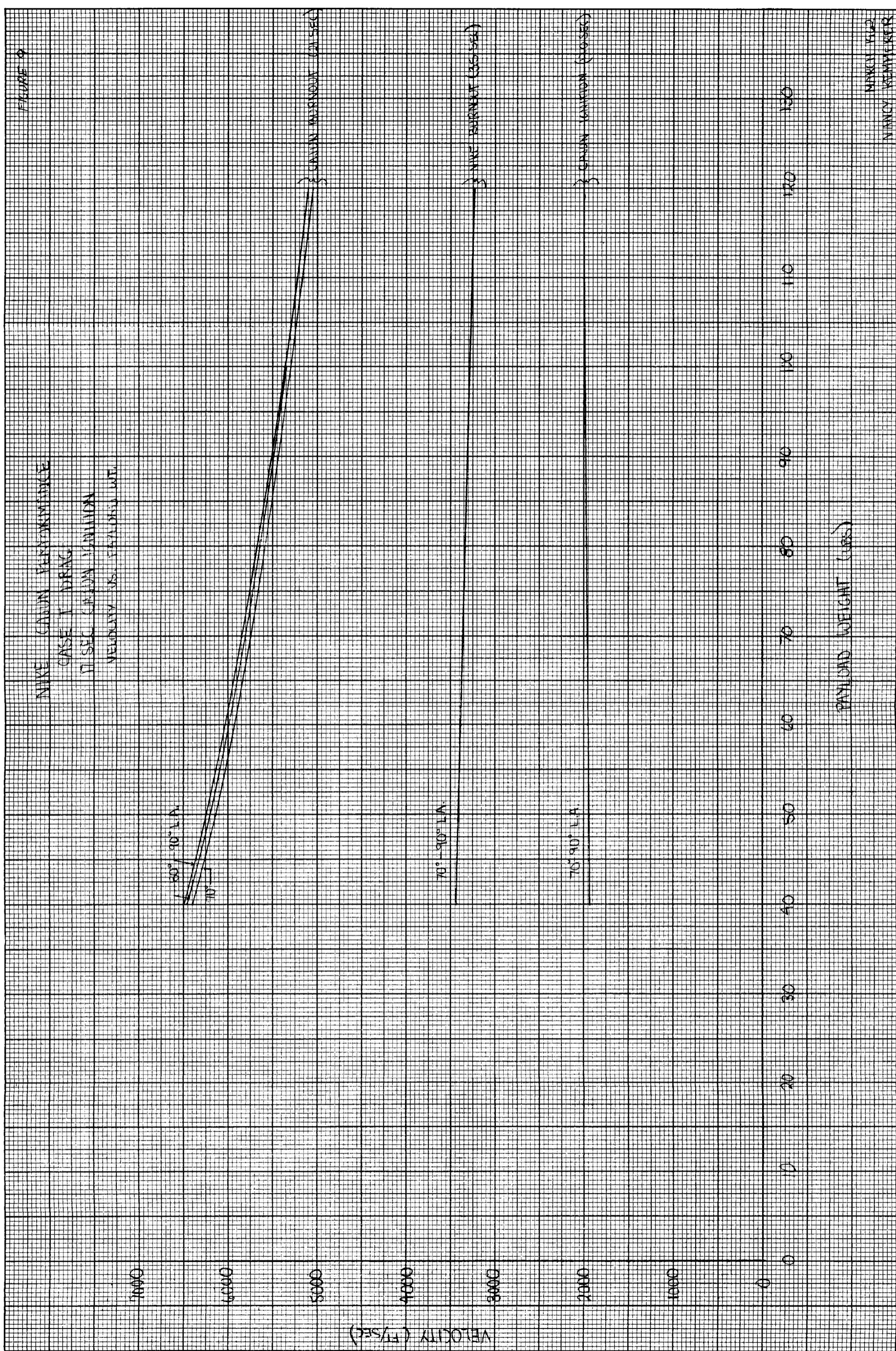
CAJUN P/L WT.	Θ_L (DEG)	h_{B01} (FT)	V_{B01} (FT/SEC)	H.R. _{B01} (FT)	Θ_{B01} (DEG)	H.R. _{IGN2} (FT)	V_{IGN2} (FT)	Θ_{IGN2} (DEG)	V_{IGN2} (FT/SEC)	H.R. _{B02} (FT)	Θ_{B02} (DEG)	V_{B02} (FT/SEC)	Θ_{B02} (DEG)	t_{max} (SEC)	V_{max} (FT/MI)	H.R. _{max} (ST.MI)	H.R. _i (ST.MI)		
40	70	5311	3432	2279	66.4	35,660	1911	16,634	62.0	51,032	6384	25,013	61.0	184.7	87.9	88.9	36A.9	177.9	
	75	5509	3432	1732	72.3	31,317	1919	12,705	68.9	53,676	6A32	19,163	68.2	199.2	102.5	75.4	362.4	150.8	
	80	5653	3432	1166	78.2	38,525	1925	8582	75.9	55,617	6444	12,911	75.4	209.6	113.9	54.5	412.5	109.2	
	82	5695	3432	935	80.5	38,876	1927	6891	78.7	56,182	6473	10,423	78.3	212.7	117.3	44.7	418.3	89.4	
	84	5728	3432	703	82.9	39,151	1928	5184	81.5	56,626	6A80	7894	81.2	215.1	120.0	34.1	422.9	68.2	
	86	5751	3432	469	85.3	39,347	1929	3463	84.3	56,942	6A85	5242	84.1	216.7	122.0	23.0	426.2	46.1	
	90	5769	3432	0	90.0	39,507	1929	0	90.0	57,198	6A89	0	90.0	218.1	123.5	0	428.8	0	
	50	70	5272	3408	2265	66.4	36,487	1929	16,668	61.9	50,587	6131	24,800	61.0	178.3	82.1	82.8	352.0	165.8
	75	5470	3407	1721	72.3	37,338	1936	12,730	68.9	53,198	6174	18,996	68.1	192.0	95.4	70.1	377.9	140.2	
	80	5614	3407	1158	78.2	38,543	1941	8598	75.9	55,111	6202	12,856	754	201.9	105.8	50.7	396.9	101.4	
	82	5655	3407	929	80.5	38,892	1943	6904	78.7	55,667	6210	10,330	78.3	204.7	108.9	41.4	402.5	83.0	
	84	5688	3407	698	82.9	39,168	1944	5193	81.5	56,104	6216	774	81.2	207.0	111.4	31.6	406.8	63.3	
	86	5711	3407	466	85.3	39,363	1945	347	84.3	56,417	6220	5195	84.1	208.6	113.2	21.3	409.9	42.7	
	90	5730	3407	0	90.0	39,523	1945	0	90.0	56,670	6224	0	90.0	209.9	114.6	0	412.4	0	
	60	70	5236	3384	2251	66.4	35,689	1944	16,694	61.9	50,247	5932	24,645	61.0	173.5	77.8	78.4	342.1	157.0
	75	5432	3384	1711	72.2	37,339	1951	12,748	68.9	52,829	5970	18,874	68.1	186.5	90.2	66.2	364.8	132.4	
	80	5574	3383	1151	78.1	38,541	1956	8609	75.9	54,719	5996	12,772	75.4	195.9	99.8	47.7	385.0	95.6	
	82	5616	3383	924	80.5	38,891	1957	6913	78.7	55,271	6003	10,261	78.3	198.6	102.7	39.0	390.3	78.2	
	84	5649	3383	694	82.9	39,166	1958	5206	81.5	55,704	6008	7722	81.2	200.8	105.0	29.8	394.4	59.7	
	86	5672	3383	463	85.3	39,362	1959	3974	84.3	56,014	6012	5160	84.1	202.3	106.6	20.1	397.3	40.3	
	90	5690	3383	0	90.0	39,521	1959	0	90.0	56,263	6015	0	90.0	203.5	107.9	0	399.7	0	
	70	70	5199	3360	2238	66.3	35,675	1958	16,711	61.9	49,919	5752	24,499	60.9	169.1	74.0	74.5	333.1	149.1
	75	5394	3360	1701	72.2	37,322	1964	12,760	68.8	52,475	5786	18,758	68.1	181.5	85.5	62.7	356.9	125.5	
	80	5535	3360	1145	78.1	38,522	1968	8617	75.9	54,345	5809	12,692	75.3	190.5	94.5	45.2	374.2	90.5	
	82	5577	3360	918	80.5	38,872	1969	6919	78.7	54,891	5815	10,197	78.3	193.2	97.2	37.0	379.3	74.0	
	84	5610	3360	690	82.9	39,146	1970	5204	81.5	55,319	5820	7673	81.2	195.2	99.3	28.2	383.2	56.4	
	86	5633	3360	460	85.2	39,343	1971	3477	84.3	55,626	5823	5128	84.1	196.7	100.9	19.0	386.1	38.1	
	90	5651	3360	0	90.0	39,500	1971	0	90.0	55,870	5826	0	90.0	197.8	102.1	0	388.3	0	
	80	70	5162	3337	2225	66.3	35,644	1970	16,722	61.9	49,598	5588	24,360	60.9	165.0	70.5	71.0	322.9	142.1
	75	5357	3336	1691	72.2	37,287	1975	12,768	68.8	52,129	5619	18,648	68.1	177.0	81.4	59.6	347.7	119.4	
	80	5498	3336	1138	78.1	38,487	1978	8621	75.8	53,984	5639	12,616	75.3	185.6	89.8	42.9	364.4	86.0	
	82	5539	3336	913	80.5	38,836	1979	6922	78.7	54,525	5644	10,136	78.2	188.2	92.3	35.1	369.3	70.3	
	84	5572	3336	686	82.9	39,110	1980	5207	81.5	54,949	5649	7627	81.2	190.1	94.3	26.8	373.1	53.4	
	86	5595	3336	458	85.2	39,306	1981	3478	84.3	55,252	5652	5097	84.1	191.5	95.8	18.1	375.8	36.2	
	90	5612	3336	0	90.0	39,461	1981	0	90.0	55,494	5654	0	90.0	192.7	96.9	0	378.0	0	

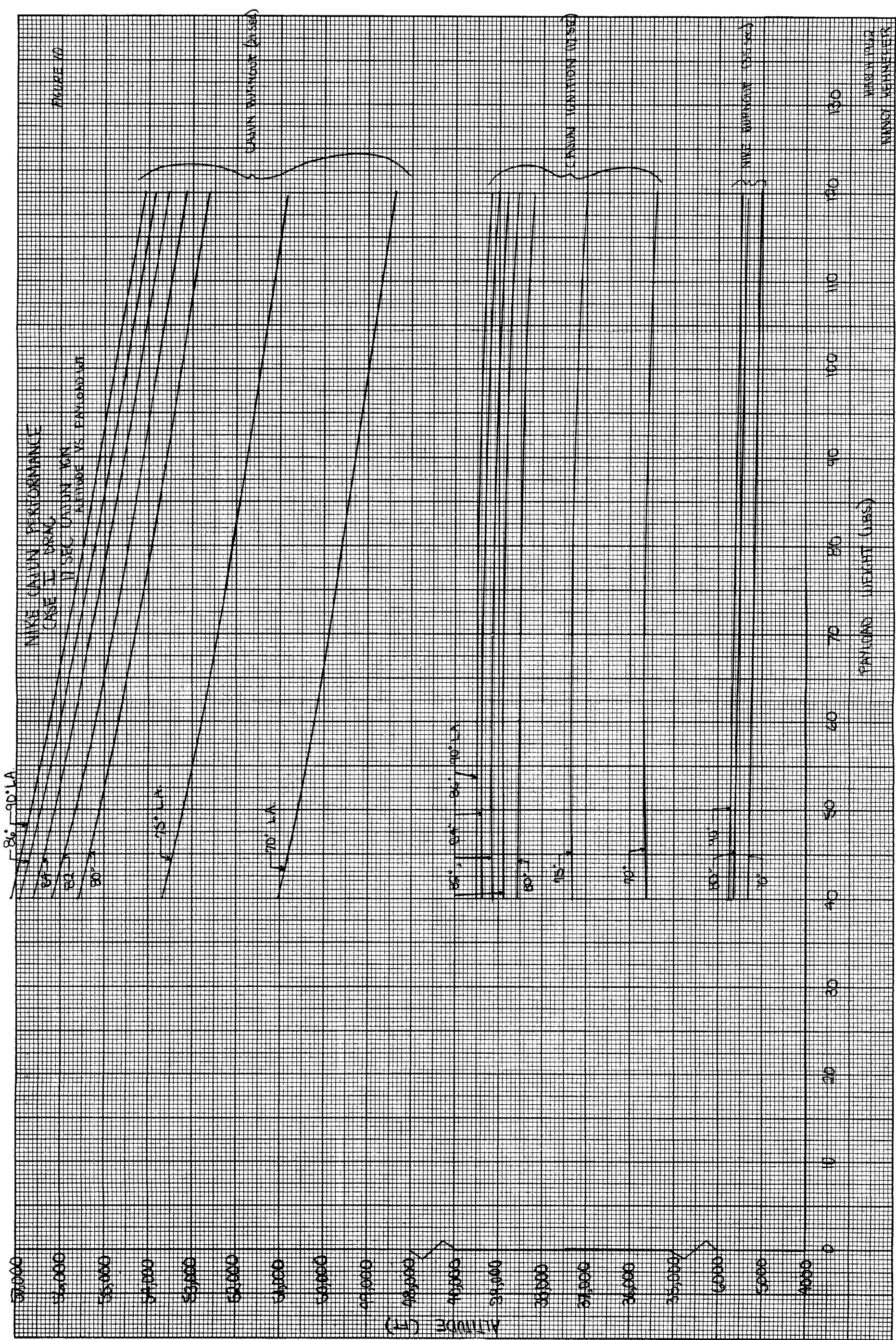
TABLE 2

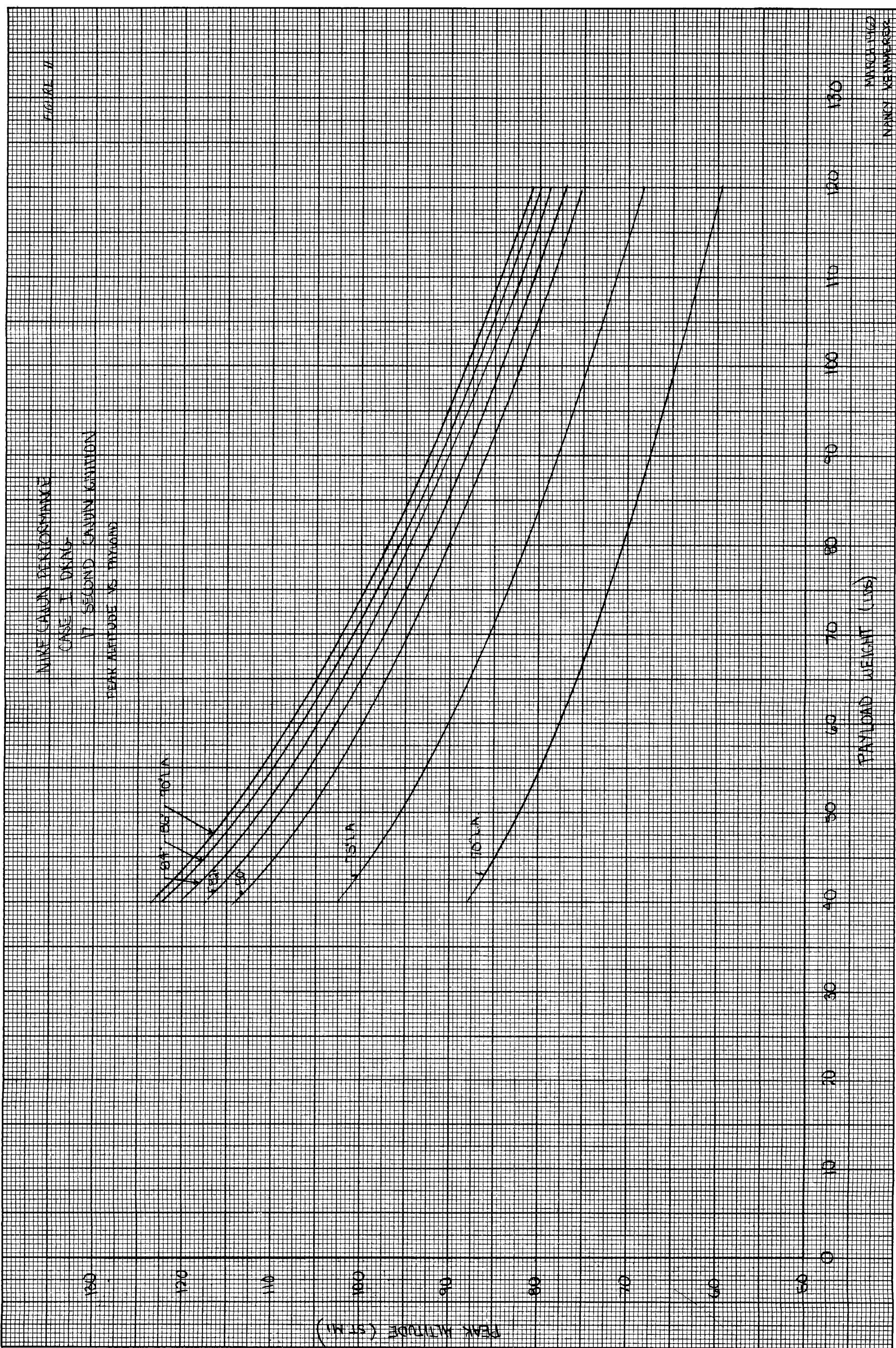
THEORETICAL NIKE CAJUN - PERFORMANCE

CASE I

CAJUN PILOT	θ_L (DEC)	h_{801} (FT)	V_{B01} (FT/SEC)	$H.R._{801}$ (FT)	θ_{801} (DEC)	V_{C02} (FT)	$H.R._{C02}$ (FT)	θ_{C02} (DEC)	t_{MAX} (SEC)	h_{MAX} (ST. M)	$H.R._{MAX}$ (ST. M)	t_{i1} (SEC)	H.R. i (ST. M)
90	70												
	75												
	80												
	82												
	84												
	86												
	90												
100	70	5091	3291	2199	66.3	35.544	1989	16,726	61.8	48,981	5298	24,100	60.8
	75	5283	3291	1671	72.2	37,180	1992	12,768	68.8	51,465	5323	18,943	68.0
	80	5423	3290	1125	78.1	38,374	1995	8621	75.8	53,287	5340	12,974	75.3
	82	5465	3290	902	80.5	38,723	1995	6922	78.7	53,819	5344	10,021	78.2
	84	5496	3290	678	82.8	38,995	1996	5206	81.5	54,236	5347	75A1	81.2
	86	5519	3290	453	85.2	39,190	1996	3478	84.3	54,534	5350	5039	84.1
	90	5536	3290	0	90.0	39,345	1996	0	90.0	54,771	5352	0	90.0
110	70	5056	3269	2186	66.2	35,477	1996	16,720	61.8	48,680	5169	23,976	60.8
	75	5248	3268	1662	72.1	37,110	1999	12,763	68.8	51,146	5193	18,346	68.0
	80	5387	3268	1118	78.1	38,302	2001	8617	75.8	52,952	5207	12,408	75.3
	82	5427	3268	897	80.5	38,647	2001	6918	78.6	53,478	5211	9967	78.2
	84	5459	3268	674	82.8	38,919	2002	5203	81.5	53,880	5214	7500	81.1
	86	5481	3268	450	85.2	39,114	2002	3476	84.3	54,187	5216	5011	84.1
	90	5500	3268	0	90.0	39,270	2002	0	90.0	54,923	5218	0	90.0
120	70	5023	3247	2174	66.2	35,901	2002	16,710	61.8	48,381	5048	23,854	60.7
	75	5213	3246	1652	72.1	37,027	2004	12,753	68.7	50,826	5068	18,250	67.9
	80	5351	3246	1112	78.1	38,215	2005	8610	75.8	52,615	5082	12,392	75.2
	82	5391	3245	892	80.4	38,542	2006	6913	78.6	53,137	5085	9914	78.2
	84	5422	3245	670	82.8	38,832	2006	5199	81.5	53,596	5088	7459	81.1
	86	5445	3245	447	85.2	39,107	2006	3473	84.3	53,841	5090	4984	84.1
	90	5462	3245	0	90.0	39,181	2006	0	90.0	54,073	5091	0	90.0







NUKE CANNON PERFORMANCE

CLASS I DRILL

7 SECOND CANNON POSITION

ARMED AND HOODIE TIME VS. TANKS AND

WEAPONS

30°

60°

90°

120°

150°

180°

210°

240°

270°

300°

330°

360°

390°

420°

450°

480°

510°

540°

570°

600°

630°

660°

720°

750°

780°

810°

840°

870°

900°

930°

960°

990°

1020°

1050°

1080°

1110°

1140°

1170°

1200°

1230°

1260°

1290°

1320°

1350°

1380°

1410°

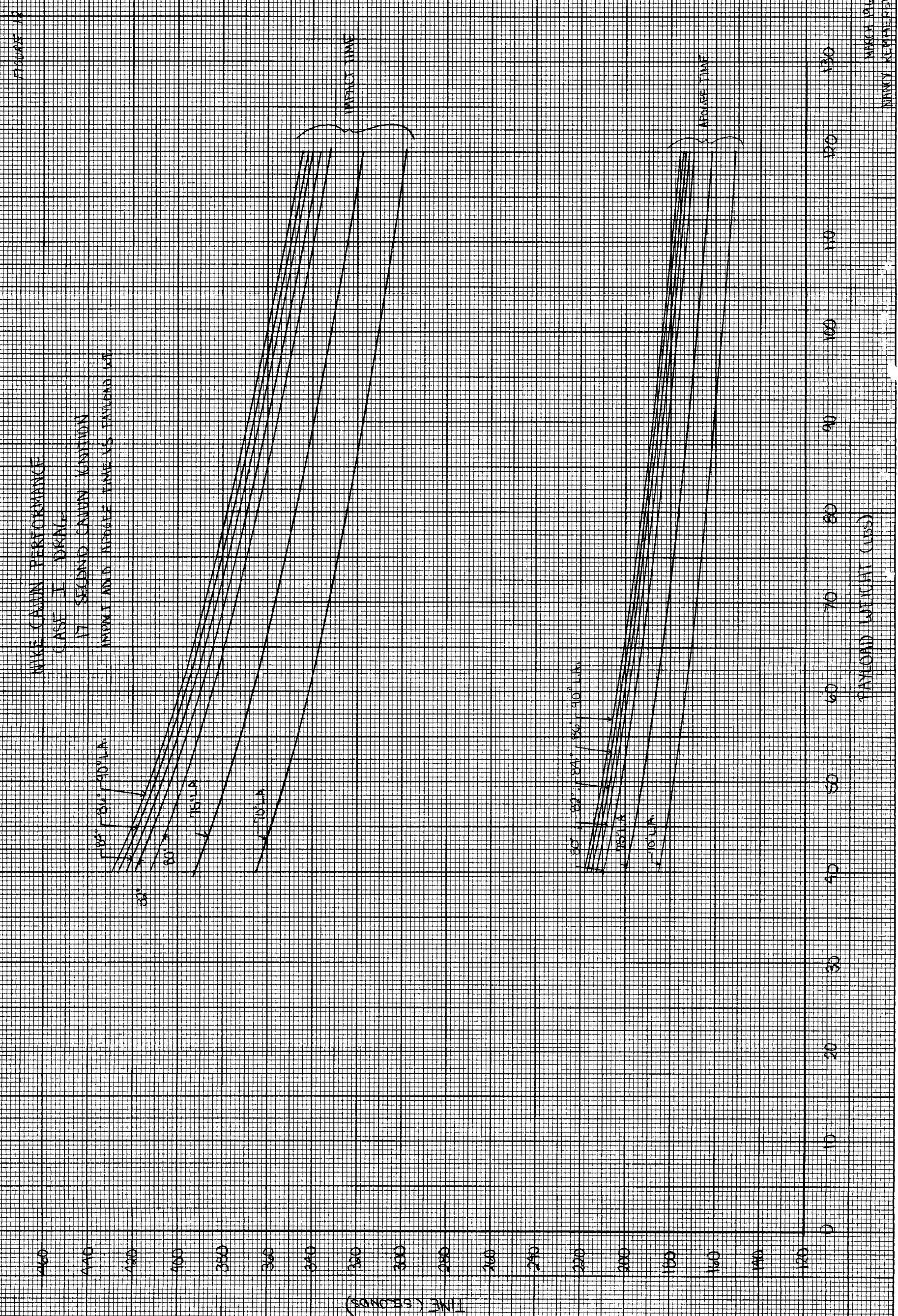
WIND TIME

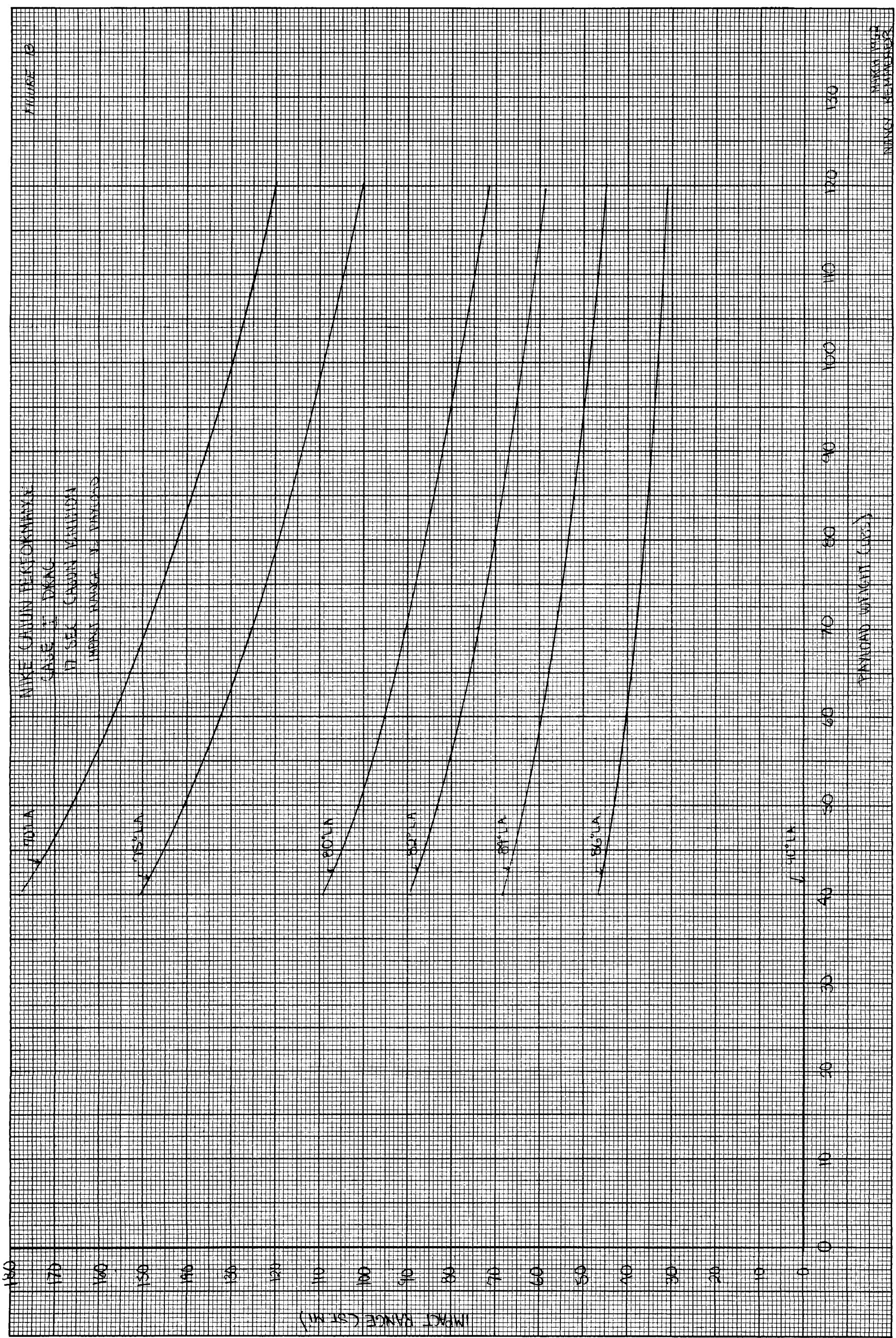
AROUSE TIME

MARK 902

MARK 903

TIME (SECONDS)





CASE II DRAG CONFIGURATION

TWO DOVAP ANTENNAE WITH 11° NOSE CONE

OR

CLEAN (NO ANTENNAE) WITH 20° NOSE CONE

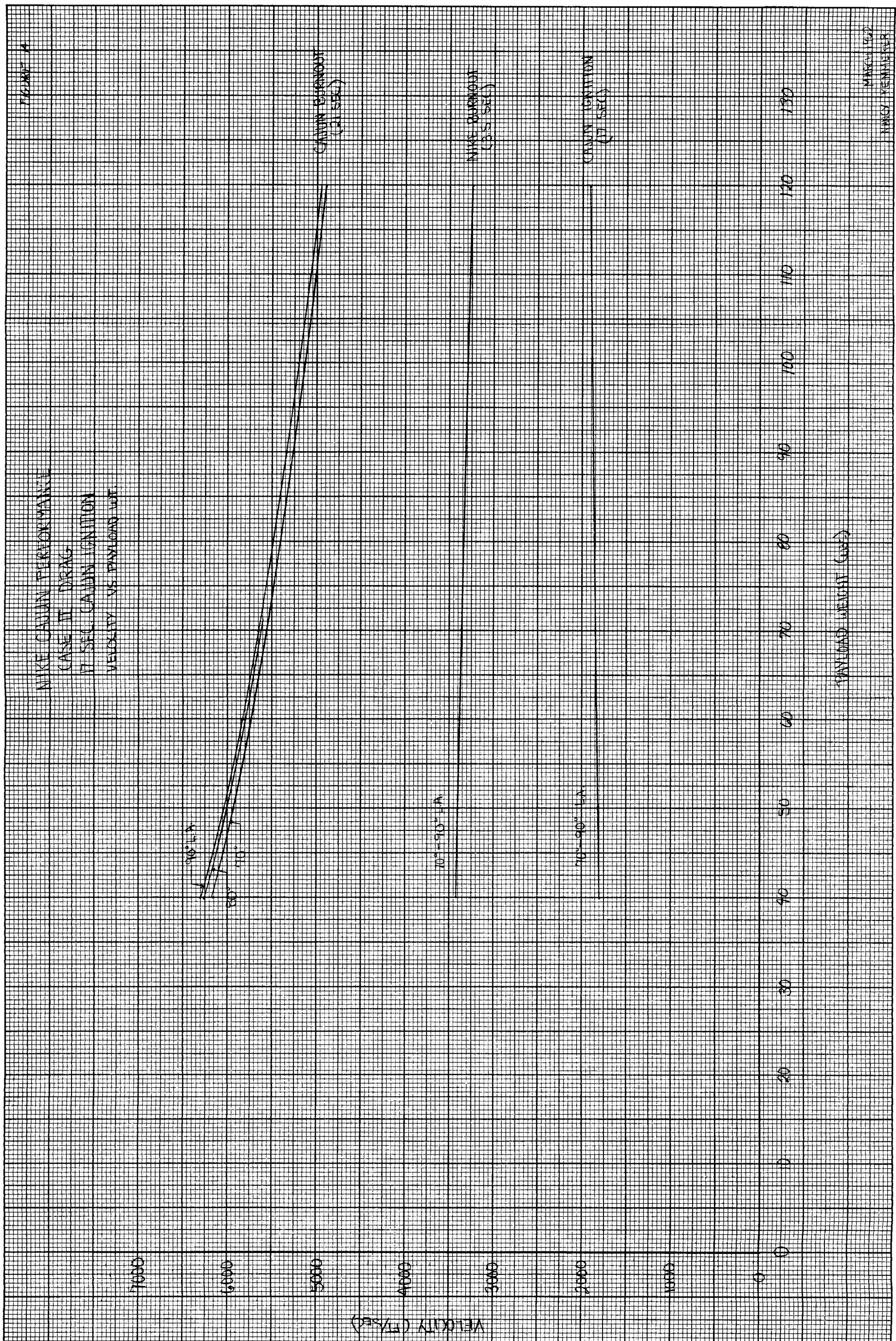
TABLE 3

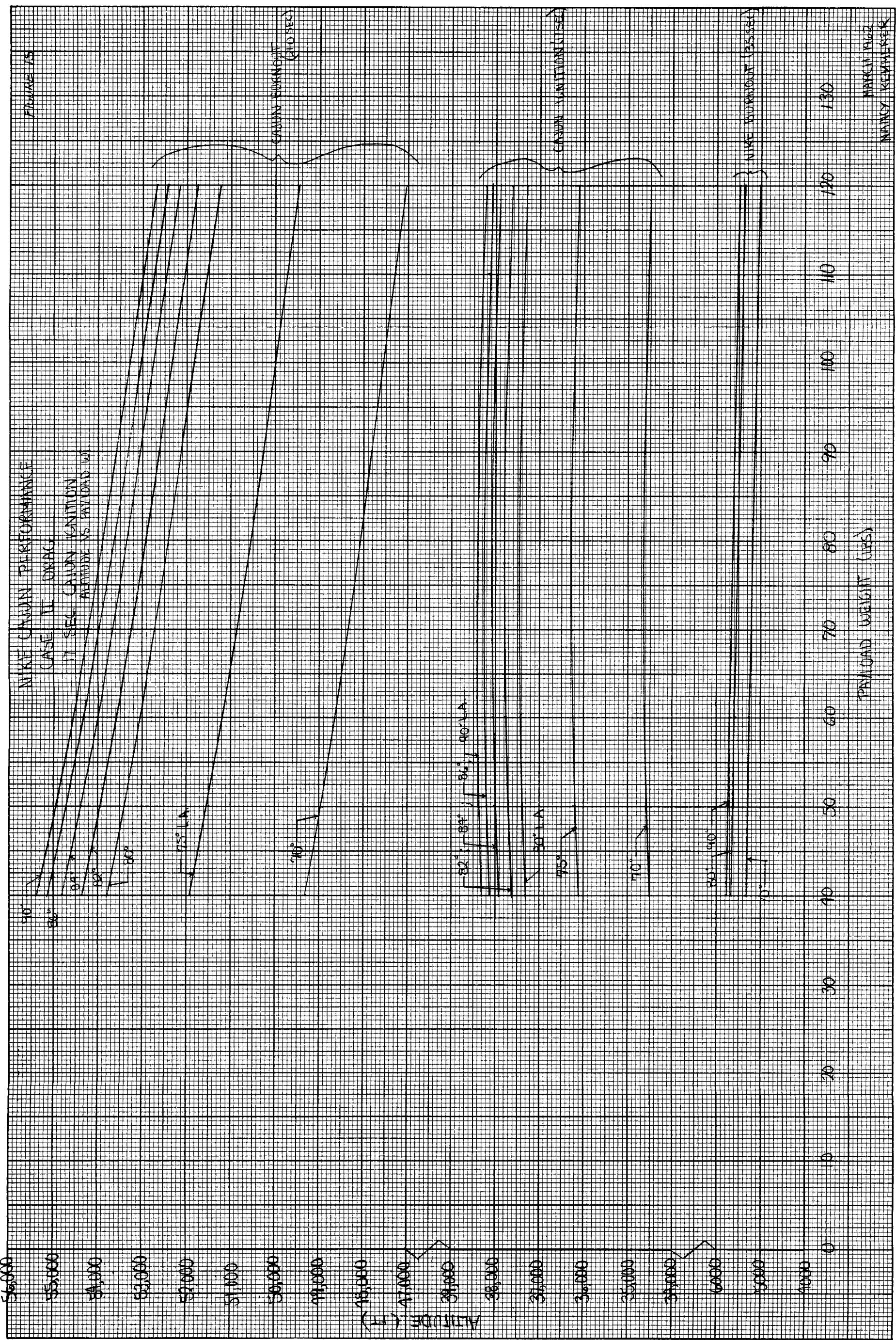
THEORETICAL NIKE CAJUN - PERFORMANCE CASE II

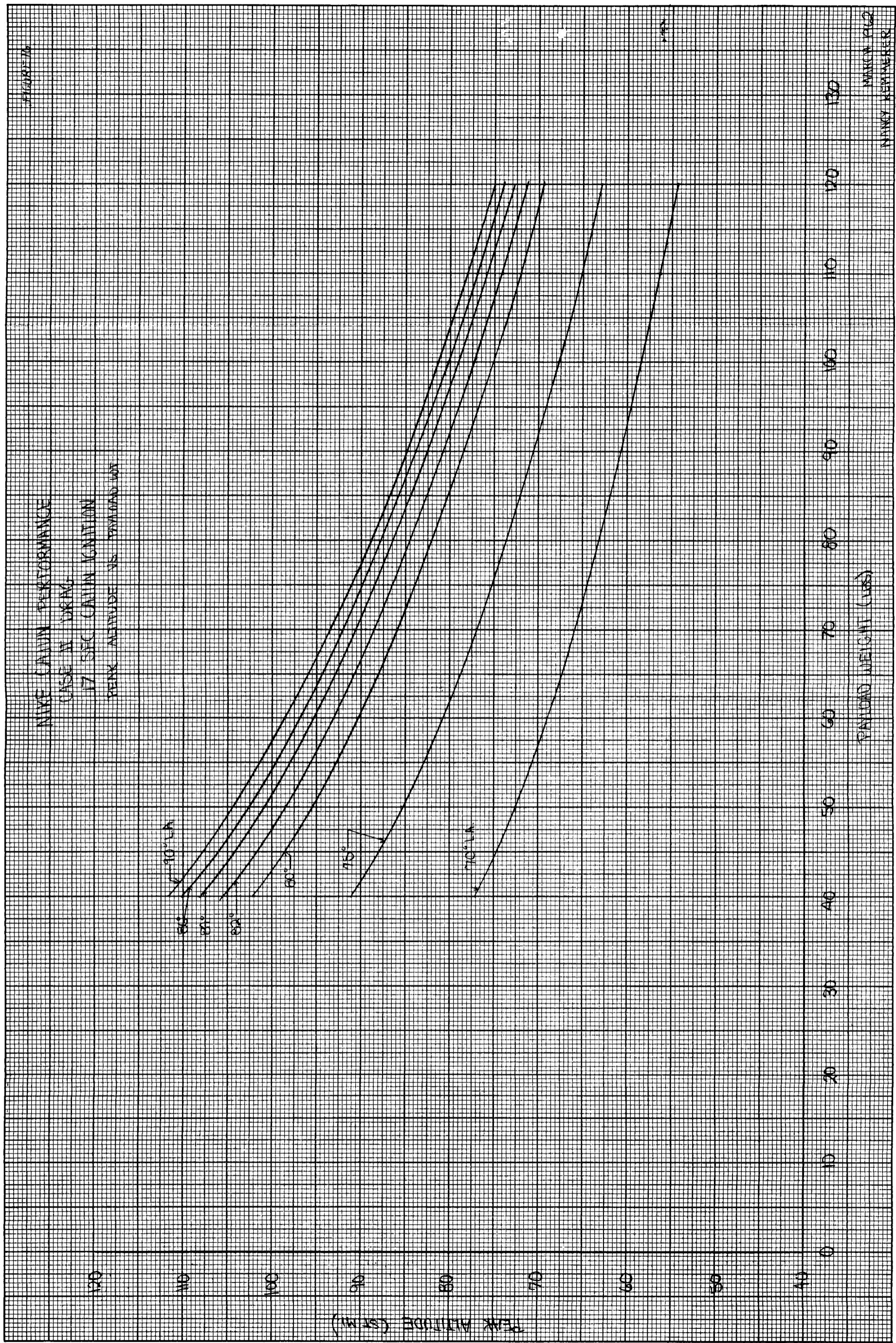
TABLE 3

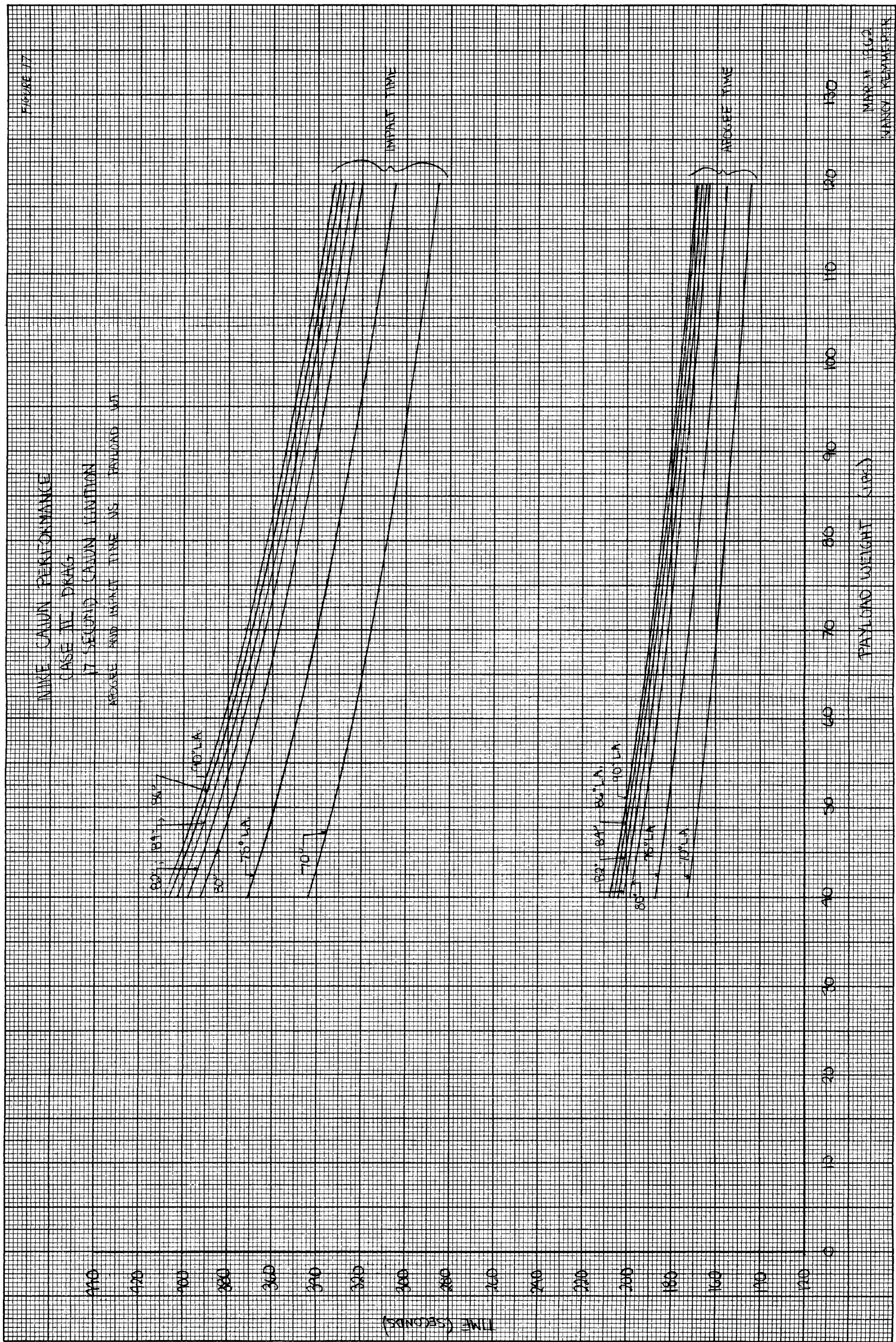
THEORETICAL NIKE CAJUN PERFORMANCE
CASE II

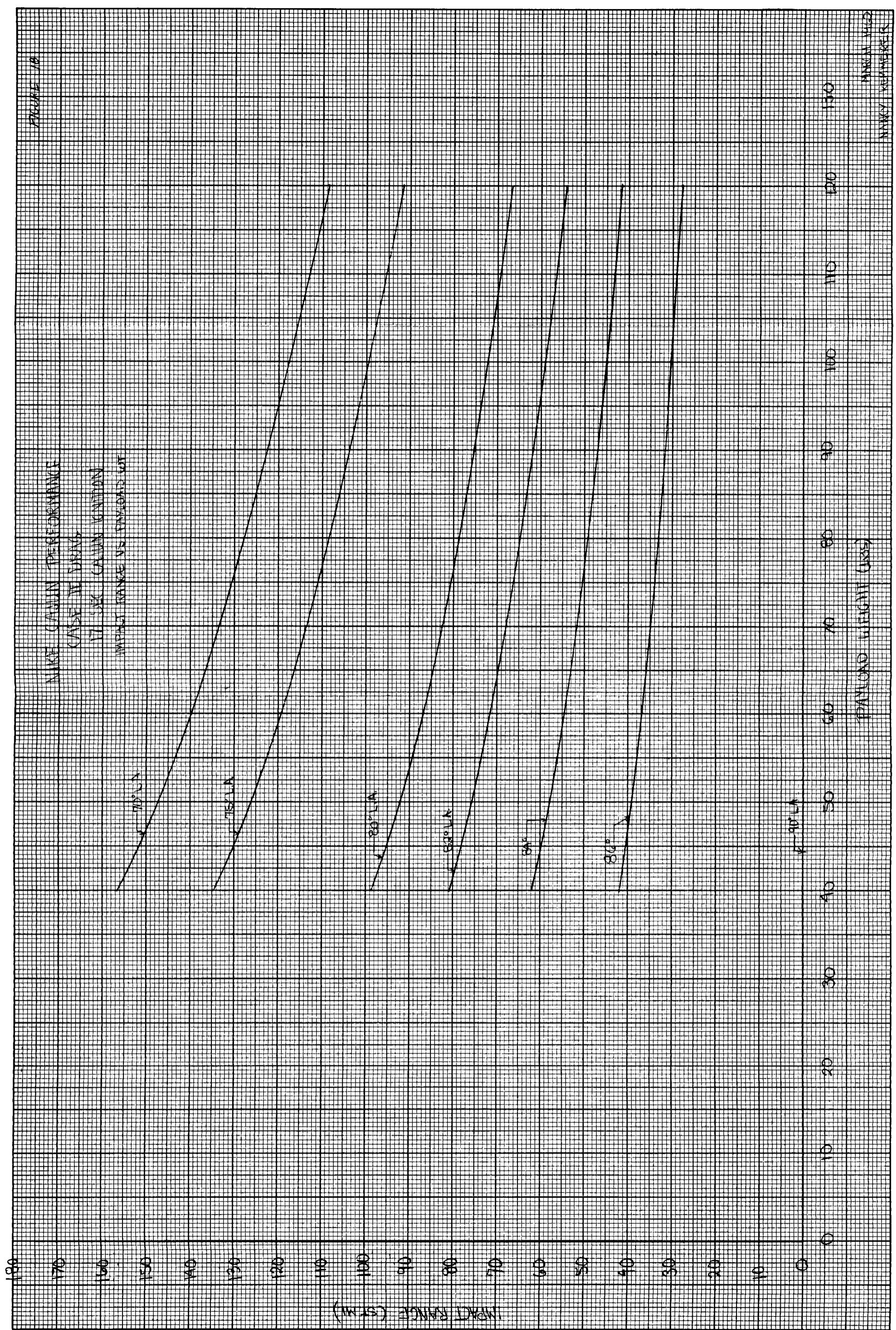
CAJUN P/L WT.	θ_c (deg.)	h_{80} (ft.)	V_{80} (ft/sec.)	$H_R\theta_{80}$ (ft)	θ_{80} (deg.)	h_{100} (ft)	V_{100} (ft/sec.)	$H_R\theta_{100}$ (ft)	θ_{100} (deg.)	V_{120} (ft/sec.)	$H_R\theta_{120}$ (ft)	θ_{120} (deg.)	t_{120} max (sec.)	$H_R t_{120}$ max (sec.)	$H_R t_{120}$ min (sec.)	
• 90	70	5246	3314	2212	66.3	34,637	1864	16,287	61.7	47,822	5265	23,579	60.6	152.9	61.2	302.7 122.4
	75	5321	3313	1681	72.2	36,251	1871	12,440	68.7	50,299	5299	18,060	67.9	164.6	70.6	51.9 103.8
	80	5461	3313	1131	78.1	37,929	1875	8902	75.7	52,113	5321	12,222	75.2	173.2	78.3	37.6 75.2
	82	5502	3313	907	80.5	37,710	1874	6747	78.6	52,641	5327	9820	78.1	175.7	80.6	30.8 61.6
	84	5534	3313	682	82.9	38,039	1877	5075	81.4	53,056	5332	7390	81.1	177.6	81.5	23.5 47.1
	86	5556	3313	456	85.2	38,231	1877	3390	84.3	53,352	5335	4939	84.1	179.0	83.8	15.9 35.2
	90	5574	3313	0	90.0	38,386	1878	0	90.0	53,592	5338	0	90.0	180.1	89.9	0 354.4
	100	5091	3291	2199	66.3	34,609	1876	16,298	61.7	47,562	5134	23,469	60.6	149.8	68.7	296.5 117.5
	75	5283	3291	1671	72.2	36,216	1881	12,446	68.7	50,011	5164	17,972	67.8	161.2	67.7	49.7 99.5
	80	5423	3290	1125	78.1	37,393	1885	8406	75.7	51,812	5187	12,161	75.2	169.5	75.0	36.0 72.0
	82	5445	3290	902	80.5	37,737	1886	6780	78.6	52,339	5190	9771	78.1	171.9	77.2	29.5 89.0
	84	5491	3290	678	82.8	38,006	1887	5077	81.4	52,751	5194	7353	81.1	173.7	79.0	22.5 45.0
	86	5519	3290	453	85.2	38,197	1887	3392	81.3	53,044	5197	4914	81.1	175.1	80.2	15.2 30.4
	90	5536	3290	0	90.0	38,351	1888	0	90.0	53,280	5200	0	90.0	176.2	81.3	0 346.9
	110	5056	3269	2186	66.2	34,546	1886	16,303	61.6	47,299	5012	23,361	60.6	147.0	66.5	290.7 113.1
	75	5248	3268	1662	72.1	36,172	1890	12,449	68.6	49,732	5039	17,887	67.8	158.0	65.1	47.8 95.6
	80	5387	3268	1118	78.1	37,347	1894	8407	75.7	51,518	5058	12,102	75.1	166.0	72.0	34.5 69.1
	82	5427	3268	897	80.5	37,690	1895	6751	78.6	52,040	5063	9724	78.1	168.3	74.1	28.3 86.6
	84	5459	3268	674	82.8	37,956	1895	5078	81.4	52,496	5067	7317	81.1	170.1	75.8	21.4 834.5
	86	5481	3268	450	85.2	38,147	1896	3392	84.3	52,738	5069	4890	84.1	171.4	77.0	14.6 337.0
	90	5500	3268	0	90.0	38,300	1896	0	90.0	52,972	5072	0	90.0	172.5	77.9	0 339.0
	120	5023	3247	2174	66.2	34,513	1894	16,303	61.6	47,037	4897	23,25A	60.5	144.3	54.4	285.1 108.9
	75	5213	3246	1652	72.1	36,116	1898	12,448	68.6	49,452	4922	17,802	67.8	154.9	62.7	45.9 91.9
	80	5351	3246	1112	78.1	37,287	1901	8406	75.7	51,221	4938	12,044	75.1	162.7	69.3	33.2 86.4
	82	5391	3245	892	80.4	37,626	1902	6750	78.5	51,736	4943	9676	78.1	164.9	71.2	27.1 54.3
	84	5422	3245	670	82.8	37,894	1902	3077	81.4	52,141	4946	7281	81.1	166.7	72.8	20.7 41.5
	86	5445	3245	447	85.2	38,087	1903	3392	84.3	52,432	4949	4866	84.0	167.9	73.9	14.0 330.0
	90	5462	3245	0	90.0	38,238	1903	0	90.0	52,663	4951	0	90.0	169.0	74.9	0 332.6











CASE III DRAG CONFIGURATION

FOUR 45° SWEPT TURNSTILE ANTENNAE WITH 11° NOSE CONE

THEORETICAL NIKE CAJUN PERFORMANCE
CASE III

TABLE 4

CAJUN P/L WT	θ_L (DEC)	V_{B0} (FT/SEC)	$H_{R, B0}$ (FT)	θ_{B0} (DEC)	V_{C0} (FT/SEC)	H_{C0} (FT)	θ_{C0} (DEC)	V_{D0} (FT/SEC)	H_{D0} (FT)	θ_{D0} (DEC)	V_{E0} (FT/SEC)	H_{E0} (FT)	θ_{E0} (DEC)	V_{F0} (FT/SEC)	H_{F0} (FT)	θ_{F0} (DEC)	t_{max} (SEC)	$V_{H, max}$ (FT/SEC)	$H_{H, max}$ (FT)	$\theta_{H, max}$ (DEC)	$t_{L, max}$ (SEC)	$V_{R, max}$ (FT/SEC)	$H_{R, max}$ (FT)	$\theta_{R, max}$ (DEC)
40	70	5311	3432	2279	66.4	33.292	1642	15536	61.5	47.329	5886	23.411	60.4	1574	64.4	65.5	65.5	319.1	130.8	130.8	65.5	319.1	130.8	130.8
	75	5509	3432	1732	72.3	34.813	1652	11,877	68.5	49.877	5951	17.962	67.7	172.8	77.6	57.7	346.8	115.2	115.2	346.8	115.2	346.8	115.2	
	80	5653	3432	1166	78.2	35.961	1659	8027	75.6	51.751	5996	12.172	75.1	183.9	88.1	42.7	367.3	85.5	85.5	367.3	85.5	367.3	85.5	
	82	5695	3432	935	86.5	36.295	1661	6447	78.5	52.297	6008	9783	78.1	187.2	91.3	35.2	373.3	70.5	70.5	373.3	70.5	373.3	70.5	
	84	5728	3432	703	82.9	36.560	1663	4850	81.4	52.728	6018	7364	81.0	189.7	93.8	27.0	378.1	54.1	54.1	378.1	54.1	378.1	54.1	
	86	5751	3432	469	85.3	36.747	1664	3240	84.2	53.025	6025	4922	84.0	191.5	95.6	18.3	381.4	36.6	36.6	381.4	36.6	381.4	36.6	
	90	5769	3432	0	90.0	36.896	1665	0	10.0	53.280	6030	0	90.0	193.0	97.1	0	384.1	0	0	384.1	0	384.1	0	
	50	70	5272	3408	2265	66.4	33.336	1666	15,603	61.5	47.015	5668	23.248	60.4	153.4	61.2	62.1	310.0	123.9	123.9	310.0	123.9	310.0	123.9
	75	5470	3407	1721	72.3	34.908	1676	11,926	68.5	49.831	5726	17.832	67.7	167.8	73.3	54.3	336.1	108.6	108.6	336.1	108.6	336.1	108.6	
	80	5614	3407	1158	78.2	36.060	1682	8060	75.6	51.379	5765	12.081	75.1	178.3	82.9	40.1	355.5	80.3	80.3	355.5	80.3	355.5	80.3	
	82	5655	3407	929	80.5	36.396	1684	6473	78.5	51.923	5777	9710	78.1	181.4	85.8	33.1	361.2	66.1	66.1	361.2	66.1	361.2	66.1	
	84	5688	3407	698	82.9	36.657	1686	4870	81.4	52.343	5785	7309	81.0	183.8	88.1	25.3	365.6	50.7	50.7	365.6	50.7	365.6	50.7	
	86	5711	3407	966	85.3	36.844	1687	3253	84.2	52.646	5791	4885	84.0	185.5	89.8	17.2	368.8	39.3	39.3	368.8	39.3	368.8	39.3	
	90	5730	3407	0	90.0	36.997	1687	0	90.0	52.841	5796	0	90.0	186.9	91.2	0	371.4	0	0	371.4	0	371.4	0	
	60	70	5236	3384	2251	66.4	33.411	1688	15,659	61.5	46.795	5497	23.139	60.4	150.4	58.9	59.6	303.0	118.0	118.0	303.0	118.0	303.0	118.0
	75	5432	3384	1711	72.2	34.985	1697	11,968	68.5	49.786	5549	17.743	67.7	164.1	70.2	51.9	328.0	103.8	103.8	328.0	103.8	328.0	103.8	
	80	5574	3383	1151	78.1	36.134	1703	8087	75.6	51.113	5585	12.019	75.1	174.1	79.1	38.2	346.5	76.5	76.5	346.5	76.5	346.5	76.5	
	82	5616	3383	924	80.5	36.471	1705	6495	78.5	51.649	5595	9660	78.0	177.0	81.8	31.5	351.9	63.0	63.0	351.9	63.0	351.9	63.0	
	84	5649	3383	694	82.9	36.736	1706	4886	81.4	52.071	5603	7271	81.0	179.3	89.0	24.1	356.2	48.2	48.2	356.2	48.2	356.2	48.2	
	86	5672	3383	463	85.3	36.922	1707	3264	84.2	52.369	5608	4860	84.0	180.9	85.5	16.3	359.2	32.7	32.7	359.2	32.7	359.2	32.7	
	90	5690	3383	0	90.0	37.074	1708	0	90.0	52.612	5612	0	90.0	182.2	86.8	0	361.7	0	0	361.7	0	361.7	0	
	70	70	5199	3360	2238	66.4	33.464	1708	15,707	61.5	46.578	5341	23.036	60.4	147.5	56.7	57.3	296.5	114.4	114.4	296.5	114.4	296.5	114.4
	75	5394	3360	1701	72.2	35.040	1711	12,003	68.5	49.047	5388	17.661	67.7	160.6	67.3	49.7	320.5	99.4	99.4	320.5	99.4	320.5	99.4	
	80	5585	3360	1145	78.1	36.188	1722	8110	75.6	50.855	5420	11,962	75.1	170.2	75.6	36.5	338.2	73.1	73.1	338.2	73.1	338.2	73.1	
	82	5577	3360	918	80.5	36.524	1723	6513	78.5	51.386	5429	9613	78.0	173.0	78.2	30.0	348.4	60.1	60.1	348.4	60.1	348.4	60.1	
	84	5610	3360	690	82.9	36.789	1725	4900	81.4	51.803	5436	7235	81.0	175.2	80.2	23.0	347.5	46.0	46.0	347.5	46.0	347.5	46.0	
	86	5633	3360	461	85.2	36.978	1726	3274	84.2	52.102	5441	4836	84.0	176.7	81.7	15.6	350.4	31.1	31.1	350.4	31.1	350.4	31.1	
	90	5651	3360	0	90.0	37.128	1726	0	90.0	52.337	5445	0	90.0	178.0	82.8	0	352.8	0	0	352.8	0	352.8	0	
	80	70	5162	3337	2225	66.3	33.464	1726	15,746	61.5	46.578	5197	22,939	60.4	149.8	54.7	55.1	290.5	110.1	110.1	290.5	110.1	290.5	110.1
	75	5357	3336	1691	72.2	35.075	1733	12,032	68.5	49.811	5291	17.583	67.7	157.4	64.7	47.7	313.6	95.3	95.					

TABLE 4

THEORETICAL NIKE CAJUN PERFORMANCE
CASE III

CAJUN P/L W/T	θ_a (deg.)	θ_{roll} (deg.)	V_{B01} (ft/sec.)	$H.R_{B01}$ (ft)	θ_{roll} (deg.)	V_{C02} (ft/sec.)	$H.R_{C02}$ (ft)	θ_{roll} (deg.)	V_{D03} (ft/sec.)	$H.R_{D03}$ (ft)	θ_{roll} (deg.)	V_{E04} (ft/sec.)	$H.R_{E04}$ (ft)	t_{max} (sec.)	V_{F05} (ft/sec.)	$H.R_{F05}$ (ft)	t_i (sec.)	$H.R_i$ (ft/mi)
90	70	5126	3314	2212	66.3	33,519	1742	15,779	61.5	46,150	5065	22,845	60.3	142.3	52.8	58.1	284.9	106.1
75	5321	3313	1681	72.2	35,094	1749	12,056	68.5	48,579	5104	17,508	67.6	154.3	62.2	45.8	307.1	91.6	
80	5461	3313	1131	78.1	36,244	1753	8144	75.6	50,358	5131	11,854	75.0	163.2	69.6	33.5	323.5	67.1	
82	5502	3313	907	80.5	36,580	1755	6540	78.5	50,860	5138	9526	78.0	165.7	71.9	27.5	328.3	55.1	
84	5534	3313	682	82.9	36,890	1756	4920	81.4	51,285	5147	7169	81.0	167.7	73.7	21.1	332.1	42.2	
86	5556	3313	456	85.2	37,028	1757	3287	84.2	51,577	5148	4791	84.0	169.2	74.9	14.3	334.8	28.5	
90	5575	3313	0	90.0	37,181	1757	0	90.0	51,815	5151	0	90.0	170.3	76.0	0	337.0	0	
100	70	5091	3291	2199	66.3	33,523	1756	15,804	61.4	45,940	4942	22,755	60.3	139.8	51.1	51.3	279.7	102.5
75	5283	3291	1671	72.2	35,094	1762	12,073	68.5	48,396	9978	17,936	67.6	181.5	60.0	44.1	301.1	88.2	
80	5423	3290	1125	78.1	36,245	1767	8156	75.6	50,114	5003	11,804	75.0	160.0	67.0	32.2	317.0	69.5	
82	5465	3290	902	80.5	36,582	1768	6550	78.5	50,632	5010	9485	78.0	162.4	69.1	26.4	321.6	52.9	
84	5496	3290	678	82.8	36,894	1769	4927	81.4	51,036	5015	7138	81.0	164.4	70.8	20.2	325.2	40.5	
86	5519	3290	453	85.2	37,031	1769	3292	84.2	51,325	5018	4771	84.0	165.8	72.0	13.7	327.8	27.4	
90	5536	3290	0	90.0	37,181	1770	0	90.0	51,556	5021	0	90	166.9	73.0	0	329.9	0	
110	70	5056	3269	2186	66.2	33,512	1769	15,823	61.4	45,728	4829	22,667	60.3	137.5	49.5	49.6	274.8	99.1
75	5298	3268	1662	72.1	35,087	1775	12,087	68.5	48,121	4862	17,966	67.6	148.8	57.9	42.5	295.5	85.0	
80	5387	3268	1118	78.1	36,234	1778	8165	75.6	49,873	4884	11,755	75.0	157.0	64.6	31.0	310.8	62.1	
82	5427	3268	897	80.5	36,571	1780	6557	78.5	50,387	4890	9445	78.0	159.4	66.5	25.5	315.3	50.9	
84	5459	3268	674	82.8	36,830	1780	4932	81.3	50,764	4895	7108	81.0	161.2	68.2	19.5	318.8	38.9	
86	5481	3268	450	85.2	37,018	1781	3295	84.2	51,073	4898	4750	84.0	162.6	69.3	13.2	321.4	26.3	
90	5500	3268	0	90.0	37,169	1781	0	90.0	51,305	4900	0	90.0	163.7	70.3	0	323.4	0	
120	70	5023	3247	2174	66.2	33,493	1780	15,837	61.4	45,518	4722	22,580	60.3	135.3	47.9	48.0	270.1	95.8
75	5213	3246	1652	72.1	35,061	1785	12,096	68.5	47,868	4762	17,296	67.6	146.2	56.0	41.0	290.2	82.0	
80	5351	3246	1112	78.1	36,208	1789	8171	75.6	49,628	4772	11,706	75.0	154.1	62.3	29.9	305.0	59.8	
82	5391	3245	892	80.4	36,542	1790	6561	78.5	50,136	4777	9406	78.0	156.4	64.2	24.5	309.4	49.1	
84	5422	3245	670	82.8	36,802	1790	4935	81.3	50,530	4782	7078	81.0	158.3	65.7	18.7	312.8	37.5	
86	5445	3245	447	85.2	36,992	1791	3297	84.2	50,890	4785	4731	84.0	159.6	68.8	12.7	315.2	25.3	
90	5462	3245	0	90.0	37,141	1791	0	90.0	51,047	4787	0	90.0	160.6	67.7	0	317.2	0	

NEXT CAUTION FIRE DRAG
CASE III DRAG
17 SECOND CAUTION ON TION
VELOCITY VS PAYLOAD

7000

6000

5000

4000

3000

2000

1000

0

VELOCITY - FT/SEC

-30° LA

0° LA

-70° 90° LA

-70° 90° LA

-70° 90°

CAUTION BURNOUT CASE (continued)

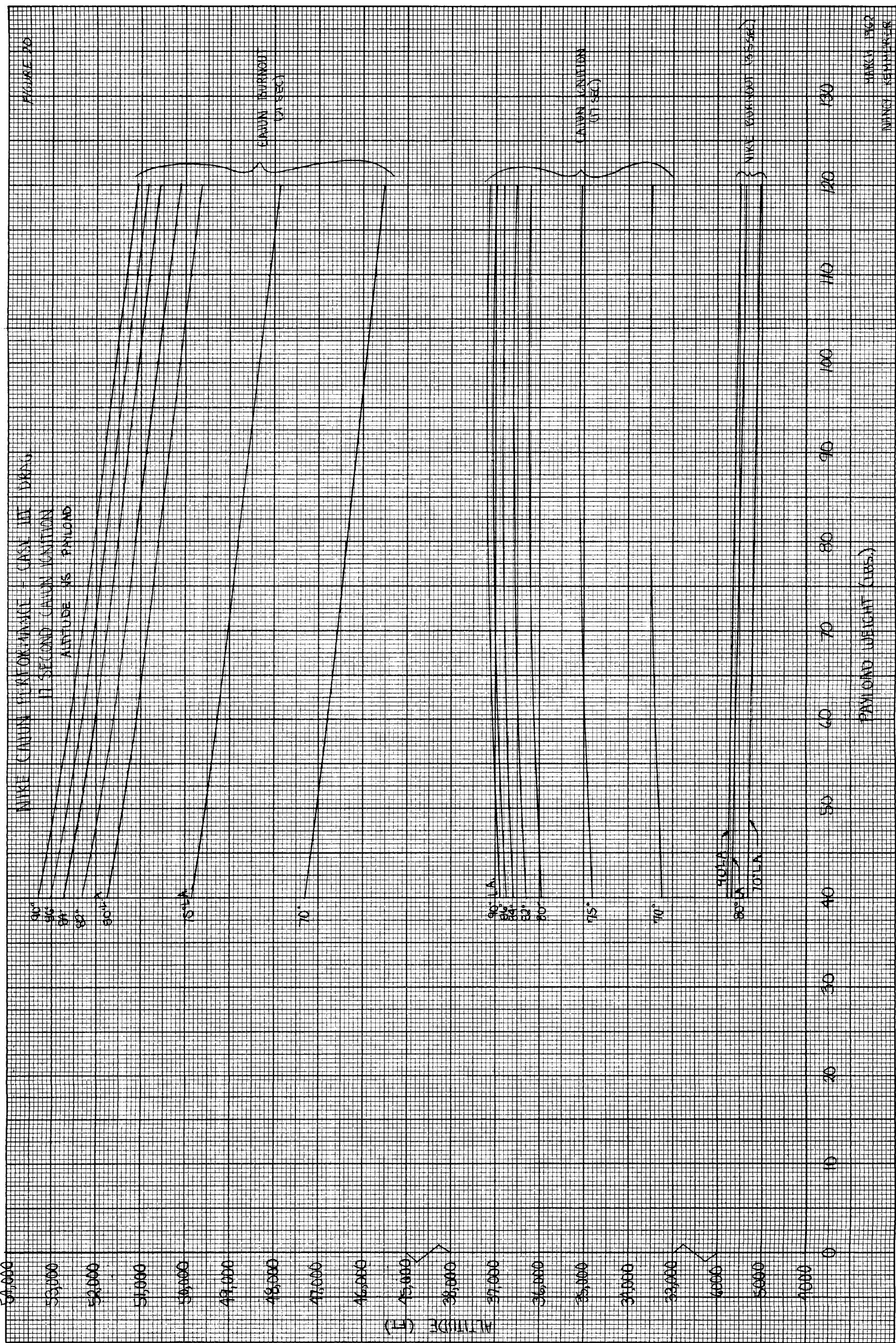
NEXT CAUTION (17 SECONDS)

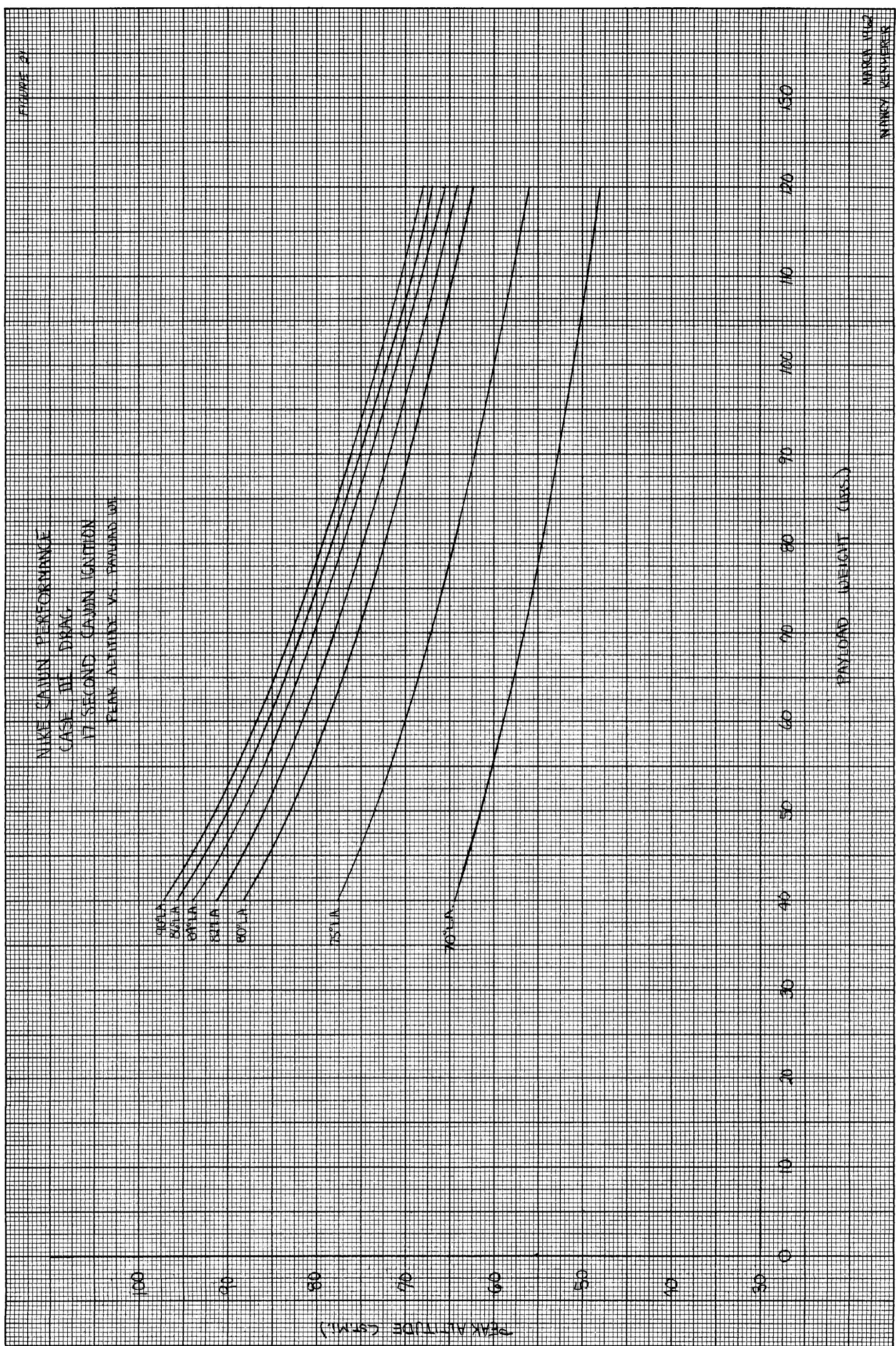
CAUTION ON TION

100
90
80
70
60
50
40
30
20
10
0

PAYOUT WEIGHT - 135

APRIL 1962
NIGHT RESEARCHER





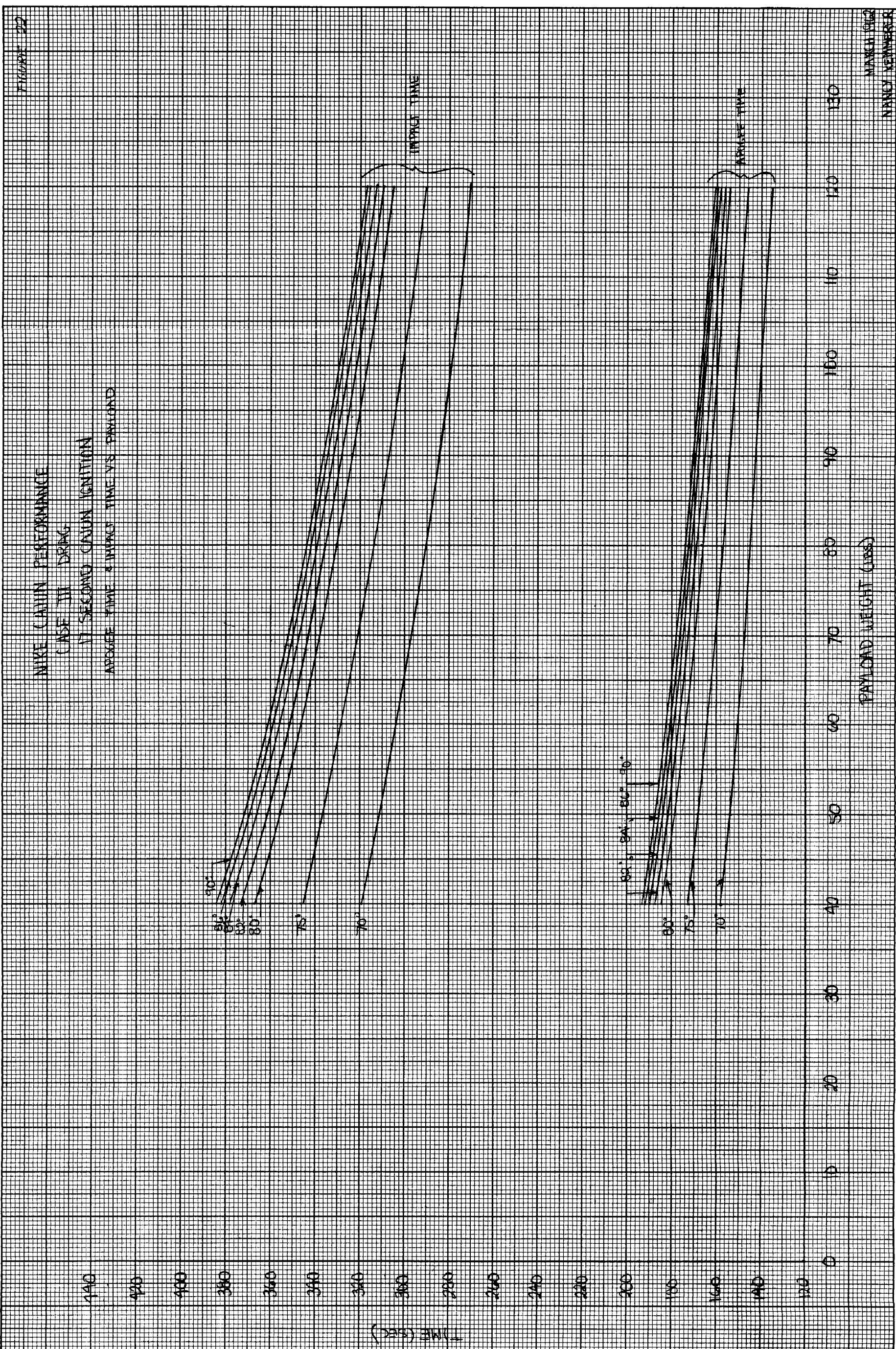
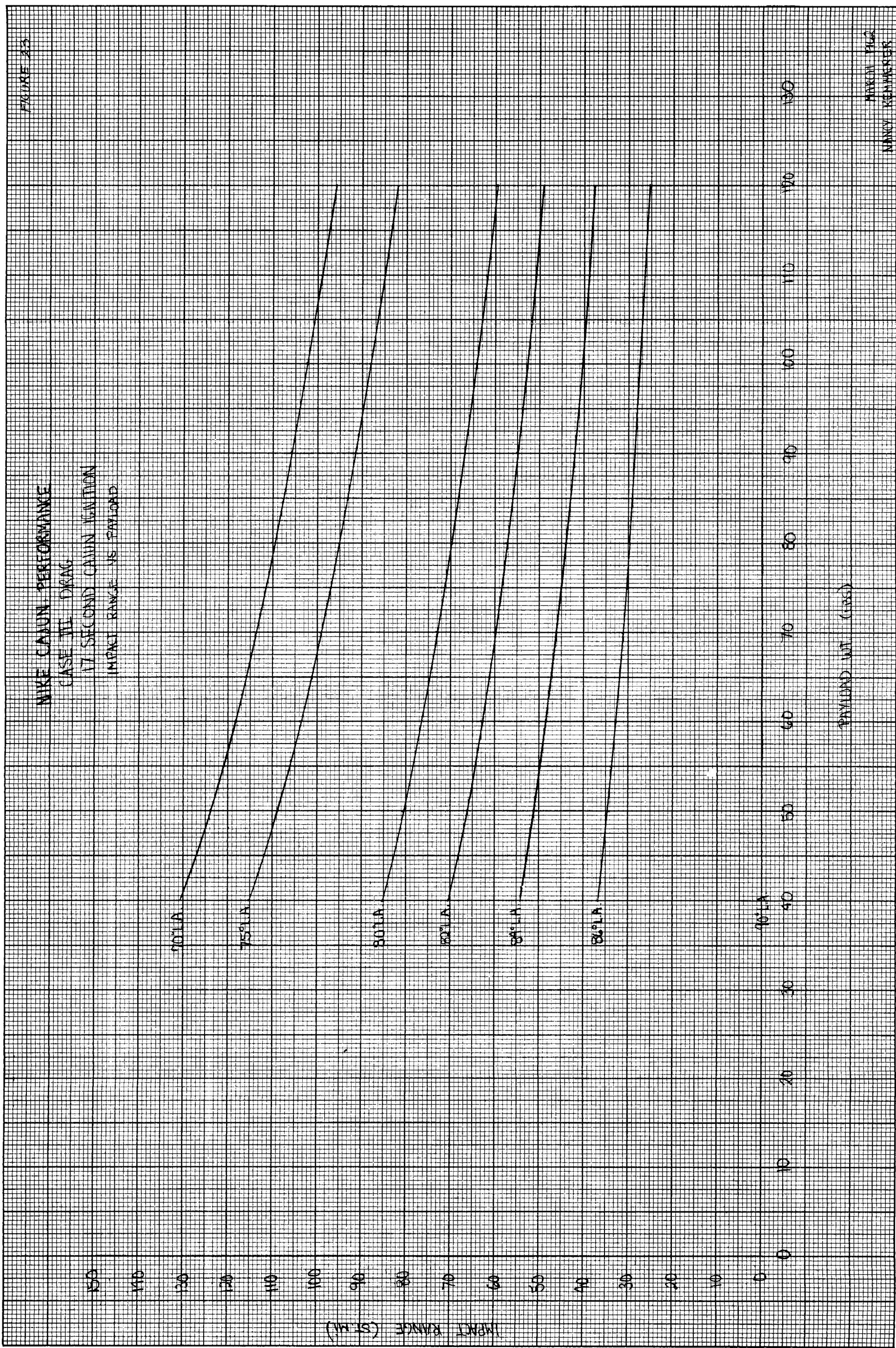


FIGURE 22



NIKE - CAJUN SOUNDING ROCKET

COMPARISON OF THEORETICAL AND ACTUAL PERFORMANCE

TABLE 5

NASA NUMBER	CAJUN DRAG CONFIGURATION	EFFECTIVE PAYLOAD WEIGHT (LBS.)	LAUNCH ELEVATION ANGLE (DEG.)	CAJUN IGNITION TIME	PEAK ALTITUDE (ST. MI.)	IMPACT RANGE (ST. MILES)		COMMENTS ON THEORETICAL
						THEORETICAL	ACTUAL	
10.25	2 QUADRALOOP 11° CONE	II	62	88	20.5	98.6	94.5 E + 4.1	19.9 20.3 E - 0.4
10.50	2-40" x $1\frac{1}{2}$ " PODS 11° CONE	II	57.4	83.5	17.3	96.2	94.6 E + 1.6	59.9 63.7 E - 3.8
10.51	2 QUADRALOOP 11° CONE	II	-58	79.5	17.5	90.7	87.8 E + 2.9	92.7 87.6 E + 5.1
10.56	2 BELL SHROUD 11° CONE	II	72	81	23.9	86.9	82.5 E + 4.4	85.2 87.0 E - 1.8
10.57	2 BELL SHROUD 11° CONE	II	72	81.3	17.0	88.0	86.8 E - 1.2	72.3 75.2 E - 2.9
10.49	4-45° TURNSTILE 11° CONE	III	60	79.1	17.6	77.8	80.7 R - 2.9	82.2 82.6 E - 0.4
10.64	4-45° TURNSTILE 11° CONE	III	71.5	79.6	17.0	74.5	81.7 R - 7.2	75.0 81.0 R - 6.0
10.74	4-45° TURNSTILE 20 CONE	III	93.5	81	17.0	69.7	62.0 R + 7.7 *	60.0 49.2 R + 108 *

R. RADAR

E. EXTRAPOLATED RADAR

* CAJUN ENCOUNTERED ROLL LOCK-IN AFTER BURNOUT BPM APR '62